

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics

## 1 How to train your solver: A method of manufactured solutions for 2 weakly-compressible smoothed particle hydrodynamics

3 Pawan Negi<sup>1</sup> and Prabhu Ramachandran<sup>1</sup>

4 Department of Aerospace Engineering, Indian Institute of Technology Bombay, Powai,  
5 Mumbai 400076

6 (\*Electronic mail: prabhu@aero.iitb.ac.in)

7 (\*Electronic mail: pawan.n@aero.iitb.ac.in)

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9 The Weakly-Compressible Smoothed Particle Hydrodynamics (WCSPH) method is a Lagrangian method that is typi-  
10 cally used for the simulation of incompressible fluids. While developing an SPH-based scheme or solver, researchers  
11 often verify their code with exact solutions, solutions from other numerical techniques, or experimental data. This  
12 typically requires a significant amount of computational effort and does not test the full capabilities of the solver. Fur-  
13 thermore, often this does not yield insights on the convergence of the solver. In this paper we introduce the method of  
14 manufactured solutions (MMS) to comprehensively test a WCSPH-based solver in a robust and efficient manner. The  
15 MMS is well established in the context of mesh-based numerical solvers. We show how the method can be applied in  
16 the context of Lagrangian WCSPH solvers to test the convergence and accuracy of the solver in two and three dimen-  
17 sions, systematically identify any problems with the solver, and test the boundary conditions in an efficient way. We  
18 demonstrate this for both a traditional WCSPH scheme as well as for some recently proposed second order convergent  
19 WCSPH schemes. Our code is open source and the results of the manuscript are reproducible.

### 20 I. INTRODUCTION

21 It has been more than four decades since the Smoothed Par-  
22 ticle Hydrodynamics (SPH) was first introduced<sup>1,2</sup>. SPH is<sup>57</sup>  
23 a meshless method and is typically implemented using La-  
24 grangian particles. The method has been applied to a wide<sup>59</sup>  
25 variety of problems<sup>3-5</sup>. However, convergence of the SPH<sup>60</sup>  
26 schemes is still considered a grand challenge problem today<sup>6</sup>.  
27 This is in part because of the Lagrangian nature of the scheme.  
28 In this paper we introduce a powerful, systematic methodol-  
29 ogy called the method of manufactured solutions<sup>7</sup> to study the  
30 accuracy and convergence of the SPH method.

31 The method of manufactured solutions<sup>7</sup> is a well estab-  
32 lished method employed in the finite volume<sup>8-10</sup> and finite ele-  
33 ment<sup>11</sup> method communities to verify the accuracy of solvers.  
34 An important part of this involves the verification of order<sup>69</sup>  
35 of convergence guarantees provided by the solver. Roache<sup>7</sup>  
36 and thereafter Salari and Knupp<sup>12</sup> formally introduced the  
37 idea of verification and validation in the context of compu-  
38 tational solvers for PDEs. Verification is a mathematical ex-  
39 ercise wherein we assess if the implementation of a numeri-  
40 cal method is consistent with the chosen governing equations.  
41 For example, verification will allow us to check whether the  
42 numerical implementation of a second-order accurate method  
43 is indeed second-order. On the other hand, validation tests  
44 whether the chosen governing equations suitably model the  
45 given physics. This is often established by comparison with  
46 the results of experiments.

47 According to Roy<sup>13</sup>, verification can be classified into two  
48 categories namely, code verification, and solution verifica-  
49 tion. In code verification, the code is tested for its correctness,  
50 whereas in solution verification, we quantify the errors in the  
51 solution obtained from a simulation. For example, in solution  
52 verification we solve a specific problem and estimate the er-  
53 ror through some means like a grid convergence study. Salari

54 and Knupp<sup>12</sup> proposed different methods for code verification  
55 viz. trend test, symmetry test, comparison test, *method of exact*  
56 *solution* (MES), and the *method of manufactured solutions*  
(MMS).

57 In the context of SPH, the comparison test and the method  
58 of exact solution are used widely to verify new schemes. In  
59 the comparison test, a solution obtained from an experiment  
60 or a well-established solver is compared with the solution ob-  
61 tained from the solver being tested. Many authors<sup>14-17</sup> use  
62 the computational results for the lid-driven cavity and flow  
63 past a cylinder problems to demonstrate the accuracy of their  
64 respective solvers. On the other hand, some authors<sup>18-20</sup> use  
65 solutions from established solvers to study the accuracy. In  
66 the MES, the exact solution of the governing equations is  
67 used to compare the accuracy as well as the order of con-  
68 vergence of the solver. For example, some authors<sup>14,15,21</sup>  
69 use the Taylor-Green vortex problem whereas others<sup>22,23</sup> use  
70 the Gresho-Chan vortex problem. We note that none of  
71 these studies have demonstrated formal second-order conver-  
72 gence for the Lagrangian Weakly-Compressible SPH (WC-  
73 SPH) scheme. Recently, Negi and Ramachandran<sup>24</sup> propose  
74 a family of second-order convergent WCSPH schemes and  
75 employ the Taylor-Green problem to demonstrate the conver-  
76 gence.

77 Despite their extensive use, the comparison and MES tests  
78 have several shortcomings<sup>12</sup>. The comparison test often re-  
79 quires a significant amount of computation since a full simu-  
80 lation for some complex problem is usually undertaken requir-  
81 ing a reasonable resolution and a large number of timesteps to  
82 attain an appropriate solution. In the case of the MES, there  
83 are very few exact solutions that exercise the full capabili-  
84 ties of the solver. For example the Taylor-Green and Gresho-  
85 Chan vortex problems are usually simulated without any solid  
86 boundaries and are only available in two-dimensions. The  
87 problems are also fairly simple and are for incompressible

fluids and this imposes additional constraints on WSPH schemes which are not truly incompressible. For example, Negi and Ramachandran<sup>24</sup> show that the error of the WSPH scheme is  $O(M^2)$ , where  $M$  is the Mach number of the flow, due to the artificial compressibility assumption. Thus the verification process requires that the WSPH solver be executed with significantly larger sound speeds than normally employed further increasing the execution time. Moreover, these methods cannot ensure that all the aspects of the solver are tested for example, it is difficult to find the order of convergence of the boundary condition implementation.

The method of manufactured solutions does not suffer from these shortcomings and is considered a state-of-the-art method for the verification of computational codes. However, this method has to our knowledge not been used in the context of the SPH thus far. In the MMS, a solution  $u = \phi(x, y, z, t)$  is manufactured such that it is sufficiently complex and satisfies some desirable properties<sup>12</sup>. We discuss these properties in detail in a later section (see section IV). Let the governing equation be given by

$$\mathcal{F}u = g, \quad (1)$$

where  $\mathcal{F}$  is the differential operator,  $u$  is the variable and  $g$  is the source term. We subject the *Manufactured Solution* (MS)  $u = \phi(x, y, z, t)$  to the governing differential equation in eq. (1). Since  $\phi$  may not be the solution of the governing equation, we obtain a residual,

$$r = \mathcal{F}\phi - g. \quad (2)$$

We add the residual  $r$  as a source term to the governing equation therefore, the modified equation is given by

$$\mathcal{F}u = g + r. \quad (3)$$

We then solve the problem along with this additional source term added to the solver. If the solver is correct we should obtain the MS,  $u$ , as the solution. We add the source term to each particle directly and this does not change the solver in any other way. The convergence of the solver may be computed numerically by solving the problem at different resolutions and finding the error in the solution.

The MMS is therefore an elegant yet simple technique to test the accuracy of a solver without making changes to the solver or the scheme. The only requirement is that it be possible to add an arbitrary source term to a particular equation. It is easy to see that the method can be applied in arbitrary dimensions. Further, we may use this technique to also test boundary conditions. By employing a carefully chosen MS one may use the method to identify specific problems with certain discretizations. For example, one may choose an inviscid solution to test only the pressure gradient term in the momentum equation. This makes it easy to discover issues in the implementation.

In Feng *et al.*<sup>25</sup> the authors use an MMS to verify their SPH implementation. However, the particles do not move and therefore it is no different than a traditional application of MMS in mesh-based methods. As mentioned earlier, the MMS has not to our knowledge been applied in the context of

the Lagrangian SPH method in order to study its accuracy. It is not entirely clear why this is the case but we conjecture that this is because the SPH method is Lagrangian and the traditional MMS has been applied in the case of traditional finite volume and finite element methods. When the particles move, it becomes difficult to satisfy the boundary conditions and have the particles moving in an arbitrary fashion. However, these issues can be handled in the context of an SPH scheme since it is possible to add and remove particles into a simulation. The lack of second order convergent SPH schemes is also a possible reason for the lack of adoption of the MMS in the SPH community. In the present work we use the recently proposed second-order convergent Lagrangian SPH schemes<sup>24</sup> to demonstrate the method. We observe that in the present work, all the schemes we consider employ some form of particle shifting<sup>15,17,26,27</sup>. This is crucial since the particles can then be constrained inside a solid domain and even if the particles move, their motion is corrected by the particle shifting algorithm. We thus do not need to add or remove particles from any of our simulations.

Our major contribution in this work is to show how one can apply the MMS to carefully study the accuracy of a modern WSPH implementation. We first obtain a suitable initial particle configuration to be used in the simulation. We then systematically show the method to construct a MS for established WSPH schemes as well as the second-order schemes proposed by Negi and Ramachandran<sup>24</sup>. We show how this can be applied to any specified shape of the domain. We show how to apply the MMS in the context of both Eulerian and Lagrangian SPH schemes. We then demonstrate how the MMS can be useful to debug a solver by deliberately changing one of the equations in the second-order convergent scheme and show the MS construction such that the change is highlighted in the order of convergence plot. We then study the convergence of some commonly used implementations for the Dirichlet and Neumann boundary conditions for solids. We demonstrate that the method can be used to study convergence for extreme resolutions as well as for three dimensional cases. The proposed method is very fast as we do not require a large number of iterations to verify the convergence. It is important to note that while we focus on verification, a validation study must be performed to ensure that the physics is accurately captured by the solver.

In summary, we present a simple, efficient, and powerful method to study convergence, and perform code verification of a WSPH solver. This is very important given that the convergence of SPH schemes is still considered a grand-challenge problem<sup>6</sup>. We make our code available as open source ([https://gitlab.com/pypr/mms\\_sph](https://gitlab.com/pypr/mms_sph)) and all the results shown in our work are fully automated in the interest of reproducibility. In the next section we briefly discuss the SPH method followed by the verification techniques used in SPH. Thereafter we discuss the MMS method and how it can be applied in the context of the WSPH scheme. We then apply the method to a variety of problems.

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 3

## 198 II. THE SPH METHOD 240

199 In the present work, we discretize the domain  $\Omega$  into  
200 equally spaced points having mass  $m$  and volume  $\omega$ . We may  
201 approximate a function  $f$  at a point  $\mathbf{x}_i$  in the domain  $\Omega$  by,

$$202 \quad \langle f(\mathbf{x}_i) \rangle = \sum_j f(\mathbf{x}_j) W_{ij} \omega_j, \quad (4) \quad 242$$

203 where  $W_{ij} = W(\mathbf{x}_i - \mathbf{x}_j, h)$ , where  $W$  is the smoothing kernel  
204 and  $h$  is its support radius.  $\omega_j = m_j / \rho_j$ ,  $\rho_j = \sum_j m_j W_{ij}$  and  $m_j$   
205 is the mass of the particle. The sum  $j$  is over all the neighbor  
206 particles of the particle  $i$ .  $\rho_j$  is commonly called the *summa-*  
207 *tion density* in the SPH literature. The eq. (4) is  $O(h^2)$  accu-  
208 rate in a uniform domain with kernel having full support<sup>28,29</sup> 247  
209 In order to obtain the gradient of the function  $f$  at  $\mathbf{x}_i$  using the  
210 kernel having full support, one may use

$$211 \quad \langle \nabla f(\mathbf{x}_i) \rangle = \sum_j (f(\mathbf{x}_j) - f(\mathbf{x}_i)) \tilde{\nabla} W_{ij} \omega_j, \quad (5) \quad 250$$

212 where  $\tilde{\nabla} W_{ij} = B_i \nabla W_{ij}$ , where  $B_i$  is the Bonet-Lok correction<sup>252</sup>  
213 matrix<sup>30</sup> and where  $\nabla W_{ij}$  is the gradient of  $W_{ij}$  w.r.t.  $\mathbf{x}_i$ . In  
214 a similar manner, many authors<sup>15,29-32</sup> propose various dis-  
215 cretizations of the gradient, divergence, and Laplacian of a  
216 function; these various forms are summarized and compared  
217 in 24.

218 The SPH method can be used to solve the Weakly-  
219 Compressible SPH equation given by

$$220 \quad \frac{d\rho}{dt} = -\rho \nabla \cdot \mathbf{u}, \quad (6) \quad 255$$

$$\frac{d\mathbf{u}}{dt} = -\frac{\nabla p}{\rho} + \nu \nabla^2 \mathbf{u}, \quad 256$$

221 where  $\rho$ ,  $\mathbf{u}$ , and  $p$  are the density, velocity, and pressure of the  
222 flow, respectively, and  $\nu$  is the dynamic viscosity of the fluid.  
223 We note here that  $\rho$  is different from the summation density  $\rho$   
224 We use  $\rho_j$  to estimate the particle volume,  $\omega_j$ . The governing  
225 equations in eq. (6) are completed by linking the pressure  $p$   
226 to density  $\rho$  using an equation of state. There are many differ-  
227 ent schemes<sup>14-16,21,33</sup> that solve eq. (6). However, they all fail  
228 to show second-order convergence. Recently, Negi and Ra-  
229 machandran<sup>24</sup> performed a convergence study of various dis-  
230 cretization operators, and propose a family of second-order  
231 convergent schemes. In this paper, we use these schemes to  
232 demonstrate the new method to study convergence of SPH  
233 schemes and compare it with the *Entropically damped arti-*  
234 *ficial compressibility* (EDAC) scheme<sup>14</sup>. We summarize the  
235 schemes considered in this study as follows:

- 270 1. L-IPST-C (Lagrangian-Iterative PST-Coupled scheme)<sup>271</sup>  
272 which is a second order scheme proposed in 24, where  
273 we discretize the continuity equation as,

$$239 \quad \frac{d\rho_i}{dt} = -\rho_i \sum_j (\mathbf{u}_j - \mathbf{u}_i) \cdot \tilde{\nabla} W_{ij} \omega_j. \quad (7) \quad 272$$

We discretize the momentum equation as,

$$\frac{d\mathbf{u}_i}{dt} = -\sum_j \frac{(p_j - p_i)}{\rho_i} \tilde{\nabla} W_{ij} \omega_j + \nu \sum_j (\langle \nabla \mathbf{u} \rangle_j - \langle \nabla \mathbf{u} \rangle_i) \cdot \tilde{\nabla} W_{ij} \omega_j \quad (8)$$

where  $\tilde{\nabla} W_{ij} = B_i \nabla W_{ij}$ , where  $B_i$  is the correction ma-  
trix<sup>30</sup>, and the  $\langle \nabla \mathbf{u} \rangle_i$  is the first order consistent gradient  
approximation given by

$$\langle \nabla \mathbf{u} \rangle_i = \sum_j (\mathbf{u}_j - \mathbf{u}_i) \otimes \tilde{\nabla} W_{ij} \omega_j. \quad (9)$$

In order to complete the system, we use a linear equa-  
tion of state (EOS) where we link pressure with the fluid  
density  $\rho$  given by

$$p_i = c_o^2 (\rho_i - \rho_o), \quad (10)$$

where  $c_o$  is the artificial speed of sound and  $\rho_o$  is the re-  
ference density. We use the standard Runge-Kutta sec-  
ond order integrator for time stepping. The time step  $\Delta t$   
is set using the stability condition given by

$$\Delta t_{cfl} = 0.25 \frac{h}{c_o + U},$$

$$\Delta t_{viscous} = 0.25 \frac{h^2}{\nu},$$

$$\Delta t_{force} = 0.25 \sqrt{\frac{h}{|\mathbf{g}|}},$$

$$\Delta t = \min(\Delta t_{cfl}, \Delta t_{viscous}, \Delta t_{force}), \quad (11)$$

where  $U$  is the maximum velocity in the domain,  $\mathbf{g}$  is  
the magnitude of the acceleration due to gravity. For  
all over test case, we set  $c_o = 20m/s$  irrespective of the  
maximum velocity in the domain. After every ten time  
step, particle shifting is applied using iterative particle  
shifting technique (IPST) to redistribute the particle in  
order to obtain a uniform distribution. We perform first  
order Taylor-series correction for velocity, and density  
after shifting.

2. PE-IPST-C (Pressure Evolution-Iterative PST-Coupled  
scheme): This method is a variation of the L-IPST-C  
scheme where a pressure evolution equation is used in-  
stead of a continuity equation<sup>24</sup>. The pressure evolution  
equation is given by

$$\frac{dp}{dt} = -\rho c_o^2 \nabla \cdot \mathbf{u} + \nu_{edac} \nabla^2 p, \quad (12)$$

where  $\nu_{edac} = \alpha h c_o / 8$  with  $\alpha = 0.5$ . The SPH dis-  
cretization of eq. (12) is given by

$$\frac{dp_i}{dt} = -\rho_i c_o^2 \sum_j (\mathbf{u}_j - \mathbf{u}_i) \cdot \tilde{\nabla} W_{ij} \omega_j + \nu_{edac} \sum_j (\langle \nabla p \rangle_j - \langle \nabla p \rangle_i) \cdot \tilde{\nabla} W_{ij} \omega_j, \quad (13)$$

where  $\langle \nabla p \rangle_i$  is evaluated using second-order consistency approximation. Since the pressure is linked with density, we evaluate the density by inverting the linear EOS given by

$$\rho_i = \frac{p_i}{c_o^2} + \rho_o. \quad (14)$$

3. TV-C (Transport Velocity-Coupled): In this method, we start with the Arbitrary Eulerian Lagrangian SPH equation<sup>16,34</sup> given by

$$\begin{aligned} \frac{d\bar{\rho}}{dt} &= -\rho \nabla \cdot (\mathbf{u} + \delta \mathbf{u}) + \nabla \cdot (\rho \delta \mathbf{u}), \\ \frac{d\bar{\mathbf{u}}}{dt} &= -\frac{\nabla p}{\rho} + \nu \nabla^2 \mathbf{u} + \nabla \cdot (\mathbf{u} \otimes \delta \mathbf{u}) - \mathbf{u} \nabla (\delta \mathbf{u}), \end{aligned} \quad (15)$$

where  $\frac{d(\cdot)}{dt} = \frac{\partial(\cdot)}{\partial t} + (\mathbf{u} + \delta \mathbf{u}) \cdot \nabla(\cdot)$  and  $\delta \mathbf{u}$  is the shifting velocity computed using

$$\delta \mathbf{u} = -M(2h)c_o \sum_j \left[ 1 + R \left( \frac{W_{ij}}{W(\Delta s)} \right)^n \right] \nabla W_{ij} \omega_j, \quad (16)$$

where  $R = 0.24$ , and  $n = 4^{35}$ . We note that the density  $\rho$  is treated as a fluid property independent of particle positions<sup>24</sup>. The main idea is to redistribute the particles using a shifting force in the governing equations instead of performing shifting post step. All the terms in the eq. (15) are discretized using a second-order accurate formulation as done in case of the L-IPST-C scheme (for details refer to 24).

4. E-C : This is an Eulerian method proposed by Negi and Ramachandran<sup>24</sup>. The governing equations for the scheme is given by

$$\begin{aligned} \frac{\partial \rho}{\partial t} &= -\rho \nabla \cdot \mathbf{u} - \mathbf{u} \cdot \nabla \rho, \\ \frac{\partial \mathbf{u}}{\partial t} &= -\frac{\nabla p}{\rho} + \nu \nabla^2 \mathbf{u} - \mathbf{u} \cdot \nabla \mathbf{u}. \end{aligned} \quad (17)$$

A similar method was proposed by Nasar *et al.*<sup>23</sup>. However, unlike the E-C method they evaluate the density as a function of particle distribution. This assumption allowed them to set the last term in the continuity equation equal to zero. This results in an increased error in the pressure as shown in 24. All the terms in the governing equations in the eq. (17) are discretized using second order accurate formulation as done in case of L-IPST-C scheme.

5. EDAC: In this method, proposed by Ramachandran and Puri<sup>14</sup>, we employ the pressure evolution equation however, density is evaluated using summation density formulation ( $\rho = \rho$  in eq. (12)). Unlike the other methods considered above, this is not a second order accurate method. The discretization of the pressure evolution in eq. (12) is given by

$$\frac{dp_i}{dt} = \sum_j \frac{m_j p_j}{\rho_j} c_o^2 (\mathbf{u}_i - \mathbf{u}_j) \cdot \nabla W_{ij}. \quad (18)$$

The momentum equation is discretized as

$$\frac{d\mathbf{u}_i}{dt} = \frac{1}{m_i} \sum_j (V_i^2 + V_j^2) \left[ \tilde{p}_{ij} \nabla W_{ij} + \tilde{\eta}_{ij} \frac{(\mathbf{u}_i - \mathbf{u}_j)}{r_{ij}^2 + \eta h_{ij}^2} \nabla W_{ij} \cdot \mathbf{r}_{ij} \right], \quad (19)$$

where  $\tilde{p}_{ij} = \frac{\rho_i p_i + \rho_j p_j}{\rho_i + \rho_j}$ , and  $\tilde{\eta}_{ij} = \frac{2\eta_i \eta_j}{\eta_i + \eta_j}$ , where  $\eta_i = \rho_i \nu_i$ .

In the next section, we consider the standard approach employed in most SPH literature where a code verification is performed to verify the SPH method.

### III. CODE VERIFICATION IN SPH

Verification and validation of a numerical method are equally important. Verification of the accuracy and convergence of a solver is found using exact solutions, solutions from existing solvers, experimental results, or manufactured solutions. The verification can also be used to identify bugs in the solver. On the other hand, validation ensures that the governing equations are appropriate for the physics and often involves comparison with experimental results.

Verification is of two kinds: (i) *code verification*, where we test the code of the numerical solver for correctness and accuracy, and (ii) *solution verification*, where we quantify the error in a solution obtained. In this paper, we focus on the code verification techniques applied to SPH. The different techniques for code verification<sup>12</sup> are:

- Trend test: Where we use an *expert judgment* to verify the solution obtained. For example, the velocity of the vortex in a viscous periodic domain should diminish with time. If the solver shows an increase of the velocity in the domain, then there is an error in the solver.
- Symmetry test: Where we ensure that the solution obtained does not change if the domain is rotated or translated. For example, if we implement an inlet assuming the flow in the  $x$  direction, we will get an erroneous result on rotating the domain by 90 degree.
- Comparison test: Where we compare the solution obtained from the solver with the solutions from an established solver or experiment. This method has been used widely by many authors in the SPH community<sup>14-19</sup> to show the correctness of their respective works.
- Method of exact solution (MES): Where we solve a problem for which the exact solution is known. For examples, in 24 this method is applied to the Taylor-Green problem for which an exact solution is known. Some authors<sup>29,36</sup> use exact solution for 1D and 2D conduction problems to demonstrate convergence.

In the context of SPH, out of the above mentioned methods comparison test and MES are employed widely. We compare solutions for the Taylor-Green and lid-driven cavity problems which are the examples of MES and comparison test, respectively.

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 5

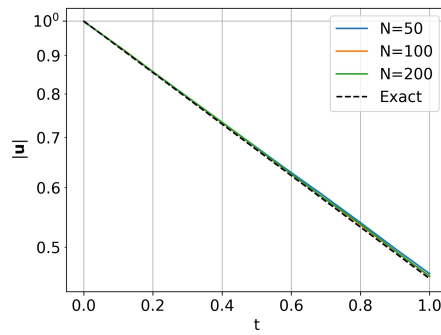
361 The Taylor-Green problem has an exact solution given by

$$\begin{aligned} 362 \quad u &= -Ue^{bt} \cos(2\pi x) \sin(2\pi y), \\ v &= Ue^{bt} \sin(2\pi x) \cos(2\pi y), \\ 363 \quad p &= -0.25U^2 e^{2bt} (\cos(4\pi x) + \cos(4\pi y)), \end{aligned} \quad (20)$$

364 where  $b = -8\pi^2/Re$ , where  $Re$  is the Reynolds number of  
365 the flow. We consider  $Re = 100$  and  $U = 1m/s$ . We solve this  
366 problem for three different resolutions viz.  $50 \times 50$ ,  $100 \times 100$ ,  
367 and  $200 \times 200$  for a two-dimensional domain of size  $1m \times 1m$   
368 for 2 sec using L-IPST-C scheme. However, we discretize the  
pressure gradient using the formulation given by

$$369 \quad \left\langle \frac{\nabla p}{\rho} \right\rangle = \sum_j \frac{(p_j + p_i)}{\rho_i} \tilde{\nabla} W_{ij} \omega_j \quad (21)$$

370 In fig. 1, we plot the decay in the velocity magnitude with

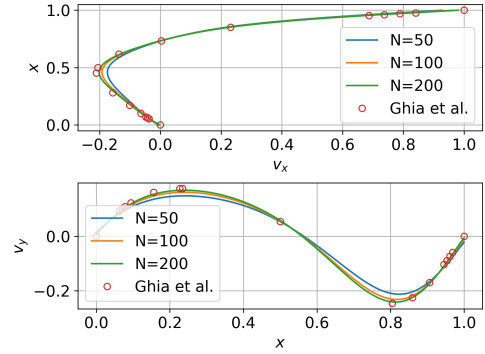


371 FIG. 1. The decay in velocity magnitude for different resolutions  
372 compared with the exact solution for the Taylor-Green problem.

373 time for different resolution compared with the exact solution.  
374 Clearly, the decay in the velocity magnitude is very close to  
375 the expected result.

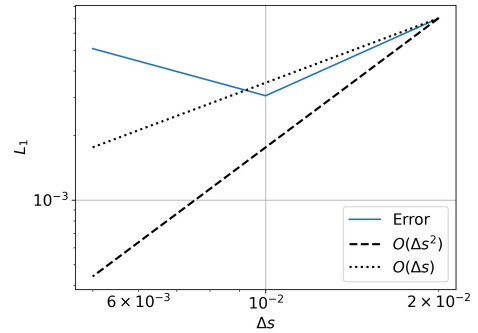
376 In the lid-driven cavity problem, we consider a two-  
377 dimensional domain of size  $1m \times 1m$  with 5 layers of ghost  
378 particles representing the solid particles. The top wall at  
379  $y = 1m$  is given a velocity  $u = 1m/s$  along the x-direction.  
380 We solve the problem using the L-IPST-C scheme for differ-  
381 ent resolution for 10 sec. However, we discretize the viscous  
382 term using the method given by Cleary and Monaghan<sup>37</sup>. In  
383 fig. 2, we plot the velocity along the centerline  $x = 0.5$  of the  
384 domain compared with the result of Ghia, Ghia, and Shin<sup>38</sup>.  
385 Clearly, the increase in resolution improves the accuracy.

386 We note that many researchers<sup>14-19</sup> use the above approach  
387 to verify their SPH schemes. Unfortunately, in both prob-  
388 lems discussed above we used a discretization which is not  
389 second-order accurate. Evidently, such kind of verification  
390 techniques are unable to detect such issues. In addition, the  
391 simulations take a significant amount of time. For example,



392 FIG. 2. The velocity along x and y direction along the center line  
393  $x = 0.5$  of the domain for the lid-driven cavity problem

394 the  $200 \times 200$  resolution lid-driven cavity case took 150 min-  
395 utes. In the case of the Taylor-Green problem since the exact  
396 solution is known one can evaluate the  $L_1$  error in velocity or  
pressure. In fig. 3, we plot the  $L_1$  error in velocity as a function



397 FIG. 3. The  $L_1$  error in velocity for the Taylor-Green problem.

of particle spacing. The  $L_1$  error is not second-order and  
diverges as we increase resolution from  $100 \times 100$  to  $200 \times 200$ .  
However, this result does not suggest to us the exact reason  
for the error.

In general, one cannot exercise specific terms in the govern-  
ing differential equation (GDE) in all the methods described  
above. Therefore, the source of error cannot be determined.  
For example, the solver may show convergence in the case  
of the Gresho-Chan vortex problem but fail for the Taylor-  
Green vortex problem due to an issue with the discretization  
of the viscous term. It is only recently<sup>39</sup> that an analytic  
solution for three dimensional Navier-Stokes equations has been  
proposed. Other recent work<sup>40</sup> has only focused on numeri-

cal investigation. It is therefore difficult to apply the MES in three dimensions. Furthermore, such studies require an ever larger computational effort. Finally, we note that the Taylor-Green vortex problem is for an incompressible fluid making it difficult to test a WCSPH scheme.

Therefore, in the context of SPH, the comparison and MES techniques are insufficient and inefficient. We require a better method to verify the solver before proceeding to validation. The method of manufactured solutions offers exactly such a technique and this is described in the next section.

#### IV. THE METHOD OF MANUFACTURED SOLUTIONS

In conventional finite volume and finite element schemes, it is mandatory to demonstrate the order of convergence and the MMS has been used for this. For the SPH method, obtaining second-order convergence has itself been a challenge until recently. Moreover, to the best of our knowledge the MMS method has not been applied in the context of SPH. In this paper, we apply the principles of MMS to formally verify SPH solvers in a fast and reliable manner. The technique facilitates a careful investigation of the various discretization operators, the boundary condition implementation, and time integrators.

In MMS, an *artificial or manufactured solution* is assumed. Let us assume the manufactured solution (MS) for  $\rho$ ,  $\mathbf{u}$ , and  $p$  in eq. (6) are  $\tilde{\rho}$ ,  $\tilde{\mathbf{u}}$ , and  $\tilde{p}$ , respectively. Since the MS is not the solution of the eq. (6), we obtain a residue,

$$\begin{aligned} s_\rho &= \frac{d\tilde{\rho}}{dt} + \tilde{\rho} \nabla \cdot \tilde{\mathbf{u}}, \\ \mathbf{s}_u &= \frac{d\tilde{\mathbf{u}}}{dt} + \frac{\nabla \tilde{p}}{\tilde{\rho}} - \nu \nabla^2 \tilde{\mathbf{u}}, \end{aligned} \quad (22)$$

where  $s_\rho$  and  $\mathbf{s}_u$  are the residue term for continuity and momentum equation, respectively. Since, we have the closed form expression for all the terms in the RHS of the eq. (22) we may introduce the residue terms as source terms in the governing equations. We write the modified governing equations as

$$\begin{aligned} \frac{d\rho}{dt} &= -\rho \nabla \cdot \mathbf{u} + s_\rho, \\ \frac{d\mathbf{u}}{dt} &= -\frac{\nabla p}{\rho} + \nu \nabla^2 \mathbf{u} + \mathbf{s}_u. \end{aligned} \quad (23)$$

Finally, we solve the eq. (23). The addition of the source terms ensures that the solution is  $\tilde{\rho}$ ,  $\tilde{\mathbf{u}}$ , and  $\tilde{p}$ .

One must take few precautions while employing the MMS:

1. The MS must be  $C^n$  smooth where  $n$  is the order of the governing equations.
2. It must exercise all the terms i.e., for any evolution equation the MS cannot be time-independent.
3. The MS must be bounded in the domain of interest. For example, the MS  $u = \tan(x)$  in the domain  $[-\pi, \pi]$  is not bounded thus, should not be used.

4. The MS should not prevent the successful completion of the code. For example, if the code assumes the solution to have positive pressure, then the MS must make sure that the pressure is not negative.

5. The MS should make sure that the solution satisfies the basic physics. For example, in a shear layer flow with discontinuous viscosity, the flux must be continuous.

We note that the MS may not be physically realistic.

We modify the basic steps for MMS proposed by Oberkampf and Roy for use in the context of WCSPH as follows:

1. Obtain the modified form of the governing equations as employed in the scheme. For example, in case of the  $\delta$ -SPH scheme, the continuity equation used is,

$$\frac{d\rho}{dt} = -\rho \nabla \cdot \mathbf{u} + D \nabla^2 \rho, \quad (24)$$

where  $D = \delta h c_o$  is the damping used, and  $\delta$  is a numerical parameter. The additional diffusive term in eq. (24) must be retained while obtaining the source term.

2. Construct the MS using analytical functions. The general form of MS is given by

$$f(x, y, t) = \phi_o + \phi(x, y, t), \quad (25)$$

where  $f$  is any property viz.  $\rho$ ,  $\mathbf{u}$ , or  $p$ ;  $\phi_o$  is a constant, and  $\phi(x, y, t)$  is a function chosen such that the five precautions listed above are satisfied.

3. Obtain the source term as done in eq. (22).

4. Add the source term in the solver appropriately. In SPH, the source term  $s = s(x, y, z, t)$ , is discretized as  $s_i = s(x_i, y_i, z_i, t)$  where subscript  $i$  denotes the  $i^{\text{th}}$  particle.

5. Solve the modified equations using the solver for different particle spacings/smoothing length ( $h$ ). The properties on the boundary particles are updated using the MS. We note that in the context of WCSPH schemes, one should not evaluate the derived quantities like gradient of velocity using the MS on the solid boundary.

6. Evaluate the discretization error for each resolution. We evaluate the error using

$$L_1(h) = \sum_j \sum_i \frac{|f(\mathbf{x}_i, t_j) - f_o(\mathbf{x}_i, t_j)|}{N} \Delta t, \quad (26)$$

where  $f$  is the property of interest,  $N$  is the total number of particles and  $\Delta t$  is the time interval between consecutive solution instances.

7. Compute the order of accuracy and determine whether the desired order is achieved.

500 The solver involves discretization of the governing equa-550  
 501 tions and appropriate implementation of the boundary con-551  
 502 ditions. The MMS can be used to determine the accuracy of552  
 503 both. However, to obtain the accuracy of boundary conditions553  
 504 the order of convergence of the governing equations should be554  
 505 at least as large as that of the boundary conditions<sup>10</sup>. Bond555  
 506 *et al.*<sup>44</sup> and Choudhary<sup>9</sup> proposed a method to construct MS556  
 507 for boundary condition verification. In order to obtain a MS  
 508 for a boundary surface given as  $F(x, y, z) = C$ , we multiply the  
 509 original MS with  $(C - F(x, y, z))^m$ . We write the new MS as 557

$$510 \quad f_{BC}(x, y, t) = \phi_0 + (C - F(x, y, z))^m \phi(x, y, t), \quad (27) 558$$

511 where  $m$  is the order of the boundary condition. For example,560  
 512 for the Dirichlet boundary  $m = 1$  and for Neumann boundary561  
 513  $m = 2$ . 562

514 In the next section, we demonstrate the application of MMS563  
 515 to obtain the order of convergence for the schemes listed in564  
 516 section II. 565

## 517 V. RESULTS 570

518 In this section, we apply the MMS to obtain the order of571  
 519 convergence of various schemes along with their boundary572  
 520 conditions. We first determine the initial particle configura-573  
 521 tion viz. unperturbed, perturbed, or packed<sup>45</sup> required for the574  
 522 MMS. We then demonstrate that one can apply the MMS to  
 523 arbitrarily-shaped domains. We then compare the EDAC and  
 524 PE-IPST-C schemes which differ in the treatment of the density.  
 525 We next apply the MMS to E-C and TV-C schemes as  
 526 they employ different governing equations compared to stan-  
 527 dard WCSPH in eq. (6). We also demonstrate the application  
 528 of the MMS method as a technique to identify mistakes in the  
 529 implementation. Finally, we employ the MMS to obtain the  
 530 order of convergence of solid wall boundary conditions. We  
 531 consider the boundary condition proposed by Maciá *et al.*<sup>46</sup>  
 532 for the demonstration. 577

533 In all our test cases, we use the quintic spline kernel with576  
 534  $h_{\Delta s} = h/\Delta s = 1.2$ , where  $\Delta s$  is the initial inter-particle spac-577  
 535 ing. We consider a domain of size  $1m \times 1m$ . We simulate all  
 536 the test cases for  $50 \times 50$ ,  $100 \times 100$ ,  $200 \times 200$ ,  $250 \times 250$ ,  
 537  $400 \times 400$ ,  $500 \times 500$ , and  $1000 \times 1000$  resolutions to obtain  
 538 the order of convergence plots. In all our simulations576  
 539 we initialize the particles using the MS. We then  
 540 solve eq. (23) and set the properties on any solid particle using  
 541 the MS before every timestep. We set a fixed time step  
 542 corresponding to the highest resolution for all the other reso-577  
 543 lutions. The appropriate time step is chosen using the criteria578  
 544 in eq. (11). We evaluate the  $L_1$  error using eq. (26) in the579  
 545 solution. 580

546 The implementation of the code for the source terms (as581  
 547 shown in eq. (22)) due to the MS are automatically gener-582  
 548 ated using the *sympy*<sup>47</sup> and *mako*<sup>48</sup> packages. We recommend583  
 549 this approach to avoid mistakes during implementation. Salari584

and Knupp<sup>12</sup> used a similar approach to automatically gener-  
 ate the source term for their solvers. We use the PySPH<sup>49</sup>  
 framework for the implementation of the schemes described in  
 this manuscript. All the figures and plots in this manuscript  
 are reproducible with a single command through the use of  
 the automan<sup>50</sup> framework. The source code is available at  
[https://gitlab.com/pypr/mms\\_sph](https://gitlab.com/pypr/mms_sph).

### A. The effect of initial particle configuration

The initial particle configuration plays a significant role in  
 the error estimation since the divergence of the velocity is cap-  
 tured accurately when the particles are uniformly arranged<sup>24</sup>.  
 In this test case, we consider three different initial configura-  
 tions of particles, widely used in SPH literature viz. unperturbed,  
 perturbed, and packed. The unperturbed configuration is the one  
 where we place the particles on a Cartesian grid such that the  
 particles are at a constant distance along the grid lines. In the  
 perturbed configuration, we perturb the particles initially placed  
 on a Cartesian grid by adding a uniformly distributed random  
 displacement as a fraction of the inter-particle spacing  $\Delta s$ .  
 For the packed configuration, we use the method proposed in<sup>24,51</sup>  
 to resettle the particles from a randomly perturbed distribution  
 to a new configuration such that the number density of the particles  
 is nearly constant. In fig. 4, we show all the initial particle  
 distributions with the solid boundary in orange.

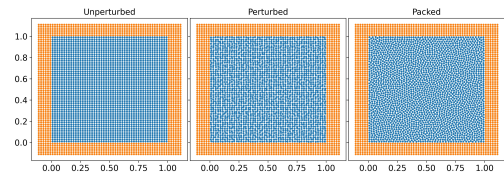


FIG. 4. The different initial particle arrangements in blue with the solid boundary in orange.

We consider the MS of the form

$$585 \quad \begin{aligned} 586 \quad u(x, y, t) &= e^{-10t} \sin(2\pi x) \cos(2\pi y) \\ 587 \quad v(x, y, t) &= -e^{-10t} \sin(2\pi y) \cos(2\pi x) \\ 588 \quad p(x, y, t) &= e^{-10t} (\cos(4\pi x) + \cos(4\pi y)) \\ 589 \quad \rho(x, y, t) &= \frac{p}{c_o^2} + \rho_o \end{aligned} \quad (28)$$

where, we set  $c_o = 20m/s$  for all our testcases. The MS complies  
 with all the required conditions discussed in section IV. We note  
 that the MS chosen resembles the exact solution of the Taylor-Green  
 problem. However, since the solver simulates the NS equation  
 using a weakly compressible formulation, we obtain additional  
 source terms when we substitute the MS to eq. (6) with  $v = 0.01m^2/s$ .  
 We obtain the source terms from the symbolic framework, *sympy*,  
 as

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 8

$$\begin{aligned}
 s_u(x, y, t) &= 2\pi u e^{-10t} \cos(2\pi x) \cos(2\pi y) - 2\pi v e^{-10t} \sin(2\pi x) \sin(2\pi y) - 10e^{-10t} \sin(2\pi x) \cos(2\pi y) + \\
 &\quad 0.08\pi^2 e^{-10t} \sin(2\pi x) \cos(2\pi y) - \frac{4\pi e^{-10t} \sin(4\pi x)}{\rho}, \\
 s_v(x, y, t) &= 2\pi u e^{-10t} \sin(2\pi x) \sin(2\pi y) - 2\pi v e^{-10t} \cos(2\pi x) \cos(2\pi y) - 0.08\pi^2 e^{-10t} \sin(2\pi y) \cos(2\pi x) + \\
 &\quad 10e^{-10t} \sin(2\pi y) \cos(2\pi x) - \frac{4\pi e^{-10t} \sin(4\pi y)}{\rho}, \\
 s_\rho(x, y, t) &= -\frac{4\pi u e^{-10t} \sin(4\pi x)}{c_0^2} - \frac{4\pi v e^{-10t} \sin(4\pi y)}{c_0^2} - \frac{10(\cos(4\pi x) + \cos(4\pi y)) e^{-10t}}{c_0^2}.
 \end{aligned} \tag{29}$$

We add  $\mathbf{s}_u = s_u \hat{\mathbf{i}} + s_v \hat{\mathbf{j}}$  to the momentum equation and  $s_\rho$  to the continuity equation as shown in eq. (23). We solve the modified WCSPH equations in eq. (23) using the L-IPST-C method for 100 timesteps where we initialize the domain using eq. (28). The values of the properties  $\mathbf{u}$ ,  $p$ , and  $\rho$  on the (orange) solid particles are set using eq. (28) at the start of every time step.

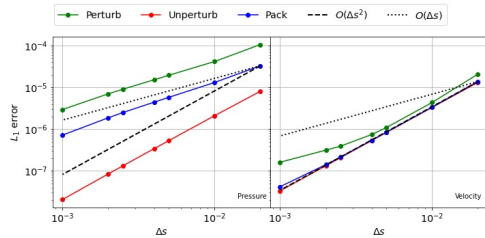


FIG. 5. The error in pressure (left) and velocity (right) with fluid particles initialized using the MS in eq. (28) and the source term in eq. (29) after 10 timesteps for the different configurations.

In fig. 5, we plot the  $L_1$  error in pressure and velocity after 10 timesteps as a function of resolution for different initial particle distributions. Clearly, the difference in initial configuration affects the error in pressure by a large amount. However, in velocity, the error is large in the case of the perturbed configuration only. The unperturbed configuration has zero divergence error at  $t = 0^{24}$ . Whereas, the perturbed configuration has high error due to the random initialization. Over the course of a few iterations, there is no significant difference between the distribution of particles for the unperturbed and the packed configurations. Therefore, we simulate the problems for 100 timesteps for a fair comparison.

In fig. 6, we plot the  $L_1$  error in pressure and velocity after 100 timesteps as a function of resolution for the cases considered. Clearly, the difference in error is reduced. However, the order of convergence is not captured accurately. This is because the initial divergence is not captured accurately by the packed and perturbed configurations. This difference can be avoided through the use of a non-solenoidal velocity field

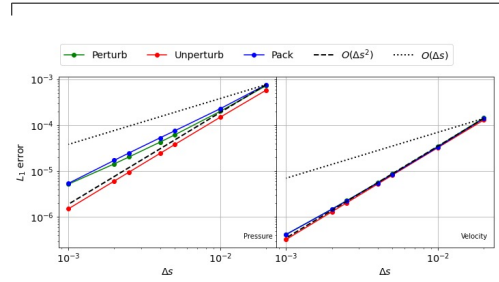


FIG. 6. The error in pressure (left) and velocity (right) with fluid particles initialized using the MS in eq. (28) and the source term in eq. (29) after 100 timesteps for all the configurations.

Therefore we consider the following modified MS,

$$\begin{aligned}
 u(x, y, t) &= y^2 e^{-10t} \sin(2\pi x) \cos(2\pi y) \\
 v(x, y, t) &= -e^{-10t} \sin(2\pi y) \cos(2\pi x) \\
 p(x, y, t) &= (\cos(4\pi x) + \cos(4\pi y)) e^{-10t}
 \end{aligned} \tag{30}$$

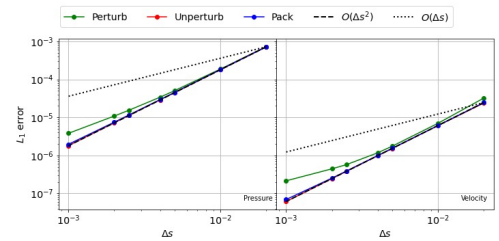


FIG. 7. The error in pressure (left) and velocity (right) with fluid particles initialized using the MS in eq. (30) and the corresponding source terms after 100 timesteps for all the configurations.

We note that the new MS velocity field is not divergence-free. We obtain the source term with  $v = 0.01 m^2/s$  as done in eq. (29). We simulate the problem by initializing the domain using MS in eq. (30). We also update the solid boundary properties using this MS before every timestep. In fig. 7, we plot the  $L_1$  error for pressure and velocity as a function



of resolution. Clearly, both the packed and unperturbed do-  
main show second-order convergence. Whereas, the perturbed  
configuration fails to show second-order convergence. There-  
fore, in the context of WCSPH schemes, one should not use  
divergence-free field in the MS. Furthermore, one should use  
either a packed or unperturbed configuration for the conver-  
gence study.

It is important to note that in stark contrast the Taylor-Green  
vortex problem the method shows second-order convergence  
irrespective of the value of  $c_o$ . In Negi and Ramachandran<sup>45</sup>  
a much higher  $c_o = 80m/s$  was necessary in order to demon-  
strate second-order convergence. Furthermore, the conver-  
gence is independent of the initial configuration after 100  
steps; therefore, we recommend simulating all the testcases  
for at least 100 timesteps to obtain the true order of conver-  
gence. It is important to note that some discretizations are  
second-order accurate when an unperturbed configuration is  
used<sup>24</sup>. In order to test the robustness of the discretization we  
recommend using a packed configuration.

### B. The selection of the domain shape

We now show the effect of the shape of the domain on the  
convergence of a scheme. We consider a square-shaped and  
butterfly-shaped domain as shown in fig. 8.

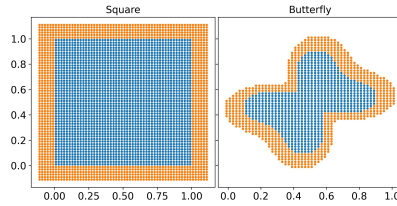


FIG. 8. The different domain shapes with solid particles in orange  
and fluid particles in blue.

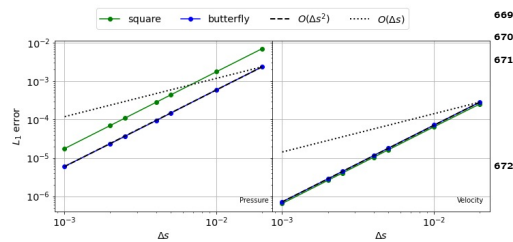


FIG. 9. The  $L_1$  error in pressure (left) and velocity (right) with in-  
crease in resolution for different shapes of the domain.

We consider the MS with the non-solenoidal velocity field  
in eq. (30) as used in the previous testcase. The source  
terms obtained remains same as before, where we consider  
 $v = 0.01m^2/s$ . We solve the modified equations using the  
L-IPST-C scheme for 100 time step for each domain. We ini-  
tialize the fluid and solid particles using the MS in eq. (30).  
We update the properties of the solid particles before every  
timestep using the same MS.

In fig. 9, we show the convergence of  $L_1$  error after 100  
timesteps in pressure and velocity as a function of resolution  
for both the domain considered. Clearly, both the domains  
considered show second-order convergence. Hence, one can  
consider any shape of the domain for the convergence study of  
WCSPH schemes using MMS. However, we only use square-  
shaped domain for all our test cases.

### C. Comparison of EDAC and PE-IPST-C

In this testcase, we compare the convergence of EDAC<sup>14</sup>  
and PE-IPST-C<sup>24</sup> schemes. These two schemes have two ma-  
jor differences. First, the discretizations used in PE-IPST-C  
method are all second-order accurate in contrast to the EDAC  
scheme. Second, the volume of the fluid given by

$$V_i = \frac{1}{\sum_j W_{ij}}, \quad (31)$$

is used in the discretization of the term  $\frac{\nabla p}{\rho}$  whereas, in PE-  
IPST-C the density  $\rho$  is independent of neighbor particle posi-  
tions. We evaluate  $\rho$  using a linear equation of state, eq. (14)

In the EDAC scheme the initial configuration of particles  
affects the results. Therefore, we consider an unperturbed  
configuration as shown in fig. 4. In order to reduce the com-  
plexity, we consider an inviscid MS given by

$$\begin{aligned} u(x, y) &= \sin(2\pi x) \cos(2\pi y) \\ v(x, y) &= -\sin(2\pi y) \cos(2\pi x) \\ p(x, y) &= \cos(4\pi x) + \cos(4\pi y). \end{aligned} \quad (32)$$

Thus, the solver must maintain the pressure and velocity fields  
in the absence of the viscosity. The source term for the EDAC  
scheme is given by

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 10

$$\begin{aligned}
 s_u(x, y) &= 2\pi u \cos(2\pi x) \cos(2\pi y) - 2\pi v \sin(2\pi x) \sin(2\pi y) - \frac{4\pi \sin(4\pi x)}{\rho} \\
 s_v(x, y) &= 2\pi u \sin(2\pi x) \sin(2\pi y) - 2\pi v \cos(2\pi x) \cos(2\pi y) - \frac{4\pi \sin(4\pi y)}{\rho} \\
 s_p(x, y) &= -1.25h(-16\pi^2 \cos(4\pi x) - 16\pi^2 \cos(4\pi y)) - 4\pi u \sin(4\pi x) - 4\pi v \sin(4\pi y).
 \end{aligned} \tag{33}$$

We note that the source term employs density  $\rho$  which is a function of particle position given by  $\frac{m_i}{V_i}$ , where  $m_i$  is the mass of the particle. In the case of the PE-IPST-C scheme, the source term is given by

$$\begin{aligned}
 s_u(x, y) &= 2\pi u \cos(2\pi x) \cos(2\pi y) - 2\pi v \sin(2\pi x) \sin(2\pi y) - \frac{4\pi \sin(4\pi x)}{\rho} \\
 s_v(x, y) &= 2\pi u \sin(2\pi x) \sin(2\pi y) - 2\pi v \cos(2\pi x) \cos(2\pi y) - \frac{4\pi \sin(4\pi y)}{\rho} \\
 s_p(x, y) &= -1.25h(-16\pi^2 \cos(4\pi x) - 16\pi^2 \cos(4\pi y)) - 4\pi u \sin(4\pi x) - 4\pi v \sin(4\pi y).
 \end{aligned} \tag{34}$$

We note that the source term  $s_p$  in eq. (33) and eq. (34) are the same. We simulate the problem with the MS in eq. (32). The (orange) solid boundary properties are reset using this MS before every time step.

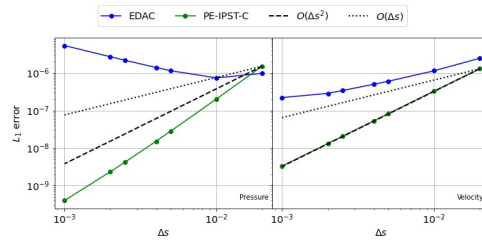


FIG. 10. The error in pressure (left) and velocity (right) with fluid particles initialized using the MS in eq. (32), and the source term in eq. (33) for EDAC and eq. (34) for PE-IPST-C after 1 timestep.

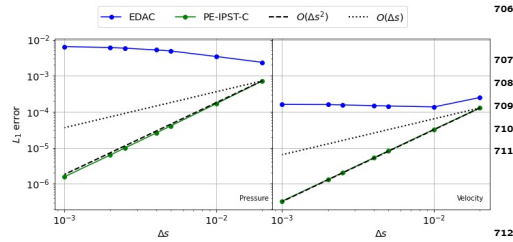


FIG. 11. The error in pressure (left) and velocity (right) with fluid particles initialized using the MS in eq. (32), and the source term in eq. (33) for EDAC and eq. (34) for PE-IPST-C after 100 timestep.

In fig. 10, we plot the  $L_1$  error in pressure and velocity after one timestep for both the schemes. Clearly, the EDAC case diverges in the case of pressure, whereas we observe a reduced order of convergence in velocity. In contrast, the PE-IPST-C scheme shows second-order convergence in velocity and higher in case of pressure. We observe this increased order only for the first iteration. In fig. 11, we plot the  $L_1$  error in pressure and velocity after 100 timesteps for both the schemes. In the case of the EDAC scheme, the order of convergence in the velocity does not remain first-order whereas, the L-IPST-C scheme shows second-order convergence in both pressure and velocity.

We note that, we use an unperturbed mesh therefore we must obtain second-order convergence to the level of discretization error for 1 timestep in the case of the EDAC scheme as well. We observe this behavior since  $\rho$  (a function of neighbor particle positions) is present in the source term which comes from the governing differential equation. Therefore, as mentioned in 24, we should treat  $\rho$  as a separate property as we do in the case of the PE-IPST-C scheme.

#### D. Comparison of E-C and TV-C

In this test case, we apply MMS to E-C and TV-C schemes introduced in section II. The governing equations for E-C scheme is given in eq. (17) whereas for TV-C in eq. (15). The expression for the source terms turns out to be same for both eq. (17) and eq. (15) governing equations given by

$$\begin{aligned}
 s_\rho &= \frac{\partial \rho}{\partial t} + \rho \nabla \cdot \mathbf{u} + \mathbf{u} \cdot \nabla \rho, \\
 s_{\mathbf{u}} &= \frac{\partial \mathbf{u}}{\partial t} + \frac{\nabla p}{\rho} - \nu \nabla^2 \mathbf{u} + \mathbf{u} \cdot \nabla \mathbf{u}.
 \end{aligned} \tag{35}$$

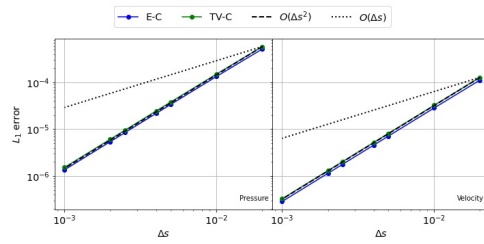
These source terms are the same as obtained in the case of the L-IPST-C scheme as well. In E-C scheme, we fix the grid and

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 11

715 add the convective term as the correction, whereas in TV-C<sub>19</sub> consider the inviscid MS in eq. (32) with the linear EOS. We do  
 716 scheme, we add the shifting velocity in the LHS of the gov<sub>20</sub> not consider the viscous term since the term introduces similar  
 717 ernaing equations. 721 error in both the schemes. We write the source term as  
 718 In order to show the convergence of the scheme, we con-

$$\begin{aligned}
 s_u(x, y) &= 2\pi u \cos(2\pi x) \cos(2\pi y) - 2\pi v \sin(2\pi x) \sin(2\pi y) - \frac{4\pi \sin(4\pi x)}{\rho}, \\
 s_v(x, y) &= 2\pi u \sin(2\pi x) \sin(2\pi y) - 2\pi v \cos(2\pi x) \cos(2\pi y) - \frac{4\pi \sin(4\pi y)}{\rho}, \\
 s_\rho(x, y) &= -\frac{4\pi u \sin(4\pi x)}{c_0^2} - \frac{4\pi v \sin(4\pi y)}{c_0^2},
 \end{aligned} \tag{36}$$

723 where  $\mathbf{s}_u = s_u \hat{\mathbf{i}} + s_v \hat{\mathbf{j}}$  is the source term for the momentum<sub>42</sub>  
 724 equation in both the schemes. We consider an unperturbed  
 725 initial particle distribution and run the simulation for 100<sub>43</sub>  
 726 timesteps. The particles are initialized with the MS in eq. (32)<sub>44</sub>  
 727 and solid boundary are reset using the MS before every time<sub>45</sub>  
 728 step. 746



748 FIG. 12. The error in pressure (left) and velocity (right) with fluid  
 749 particles initialized using the MS in eq. (32) and the source term in  
 750 eq. (36) after 100 timesteps for the different schemes. 756

729 In fig. 12, we plot the  $L_1$  error in pressure and velocity as  
 730 a function of resolution for both the schemes. Since we use  
 731 second-order accurate discretization in both the schemes, they  
 732 show second-order convergence in both pressure and velocity  
 733 as expected. Thus, we see that the modified governing equa-  
 734 tions (eq. (15) and eq. (17)) must be considered to obtain the  
 735 source term for the schemes.

### 736 E. Identification of mistakes in the implementation

737 In this section, we demonstrate the use of MS as a technique  
 738 to identify mistakes in the implementation. We use the L-<sub>757</sub>  
 739 IPST-C scheme, and introduce either erroneous or lower order<sub>758</sub>  
 740 discretization for a single term in the governing equations. We<sub>759</sub>  
 741 then use the proposed MMS to identify the problem. 760

### 1. Wrong divergence estimation

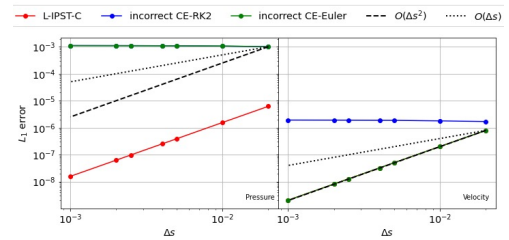
We introduce an error in the discretized form of the continuity equation used in the L-IPST-C scheme. We refer to this modified scheme as *incorrect CE*. We write the *incorrect* discretization for the divergence of velocity as

$$\langle \nabla \cdot \mathbf{u} \rangle = \sum_j (\mathbf{u}_j + \mathbf{u}_i) \cdot \bar{\nabla} W_{ij} \omega_j, \tag{37}$$

747 where the error is shown in red. Since only the continuity  
 748 equation is involved, we use the inviscid MS given by

$$\begin{aligned}
 u(x, y) &= (y-1)^2 \sin(2\pi x) \cos(2\pi y) \\
 v(x, y) &= -\sin(2\pi y) \cos(2\pi x) \\
 p(x, y) &= (y-1) (\cos(4\pi x) + \cos(4\pi y))
 \end{aligned} \tag{38}$$

751 The source terms can be determined by subjecting the above  
 752 MS to eq. (6). We simulate the problem for 1 timestep with  
 753 a packed domain (see fig. 4). In order to test erroneous or  
 754 lower order discretization in the scheme, we recommend the  
 755 simulation of only one timestep with a packed initial particle  
 756 distribution.



737 FIG. 13. The error in pressure (left) and velocity (right) with fluid  
 738 particles initialized using the MS in eq. (32) and the source term  
 739 in eq. (36) after 1 timestep for L-IPST-C and the scheme with the  
 740 divergence computed using the incorrect eq. (37).

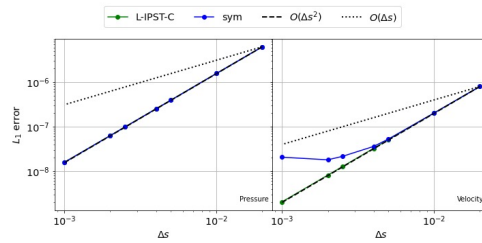
In fig. 13, we plot the  $L_1$  error in pressure and velocity as a function of the resolution for the L-IPST-C scheme and

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 12

761 its variant *incorrect CE* with two time integrators, Euler and  
 762 RK2. Clearly, the error in pressure increases by a significant  
 763 amount and the order of convergence is zero for *incorrect CE*.  
 764 However, the error in pressure propagates to velocity in case  
 765 of the RK2 integrator. Therefore, we recommend that one use  
 766 single stage integrators while using MMS as a technique to  
 767 identify mistakes. By looking at *incorrect CE-Euler* plot in  
 768 fig. 13 we can immediately infer that there is an error in either  
 769 the continuity equation or the equation of state.

## 770 2. Using a symmetric pressure gradient discretization

771 In this testcase, we use a symmetric formulation as used by  
 772 21, 24, and 52 for the pressure gradient term in the L-IPST-  
 773 C scheme. We refer to this method as *sym*. Since only the  
 774 pressure gradient is involved, we use the same MS as in the  
 775 previous case.



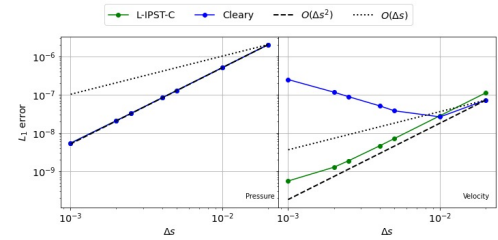
776 FIG. 14. The error in pressure (left) and velocity (right) with fluid  
 777 particles initialized using the MS in eq. (32) and the source term in  
 778 eq. (36) after 1 timestep for L-IPST-C and the scheme with pressure  
 779 gradient computed using symmetric formulation.

778 In fig. 14, we plot the  $L_1$  error after 1 timestep in pressure  
 779 and velocity as a function of resolution for L-IPST-C and *sym*  
 780 schemes. Clearly, the order of convergence is affected in the  
 781 velocity only. Therefore, it is evident that an inconsistent pres-  
 782 sure gradient discretization is used.

## 783 3. Using inconsistent discrete viscous operator

784 In this testcase, we use the formulation proposed by Cleary  
 785 and Monaghan<sup>37</sup> to approximate the viscous term in the L-  
 786 IPST-C scheme. We refer to this method as *Cleary*. Since  
 787 viscosity is involved, we use the MS involving viscous effect  
 788 given by eq. (30). While testing the viscous term we use a  
 789 high value of  $\nu = .25m^2/s$  such that the error due to viscosity  
 790 dominates the error in the momentum equation. We simulate  
 791 the problem with a packed configuration of particles for 1  
 792 timestep using the MS in eq. (30) and with the corresponding  
 793 source terms. We fix the timestep using eq. (11) such that we  
 794 satisfy the stability condition.

795 In fig. 15, we plot the  $L_1$  error in pressure and velocity as  
 796 a function of resolution for L-IPST-C and *Cleary* schemes

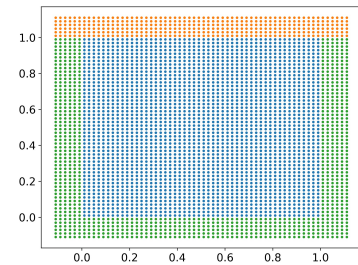


797 FIG. 15. The error in pressure (left) and velocity (right) with fluid  
 798 particles initialized using the MS in eq. (30) and the corresponding  
 799 source term after 1 timestep for L-IPST-C and the scheme with vis-  
 800 cous term discretized using formulation given by Cleary and Mon-  
 801 aghan<sup>37</sup>.

798 Since the viscous formulation by Cleary and Monaghan<sup>37</sup>  
 799 does not converge in the perturbed domain<sup>24</sup>, we observe di-  
 800 vergence in the velocity. Therefore, we infer that there is an  
 801 error in the viscous term.

## 802 F. MMS applied to boundary condition

803 In this section, we use MMS to verify the convergence of  
 804 boundary conditions in SPH. In order to do this, the scheme  
 805 used must converge at least as fast as the boundary conditions.  
 806 Therefore, we consider the second-order convergent L-IPST-  
 807 C scheme. We study the Dirichlet boundary conditions for  
 808 pressure and velocity, no-slip and slip velocity boundary con-  
 809 ditions, and the Neumann pressure boundary condition. We  
 810 consider an unperturbed domain as shown in fig. 16, where  
 811 we solve the fluid equations using the L-IPST-C scheme for  
 812 the blue particles and set the MS before every time step for  
 813 the green particles. We set the properties in the orange par-  
 814 ticles using the appropriate boundary condition we intend to  
 815 test. For example, if we set the pressure Dirichlet boundary  
 816 condition in SPH then we set velocity and density using the  
 817 MS. In order to obtain rate of convergence, we evaluate  $L_\infty$



818 FIG. 16. Different particle used for testing the boundary condition  
 819 with fluid in blue, MS solid boundary in green, and SPH solid bound-  
 820 ary in orange.

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 13

error using,

$$L_{\infty}(N) = \max\{|f(\mathbf{x}_i) - f(\mathbf{x}_o)|, i = 1, \dots, N\}, \quad (39)$$

where  $N$  is the total number of fluid particles for which  $y > 0.9$ , and  $f(\mathbf{x}_i)$  and  $f(\mathbf{x}_o)$  are the computed and exact value of the property of interest, respectively. We consider only a portion near the boundary since only that region is affected the most by the boundary implementation. In the following sections, we test the different boundary conditions in SPH using MMS.

### 1. Dirichlet boundary condition

In this testcase, we construct the MS for boundary condition as discussed in section IV. In order to set the homogenous boundary condition at  $y = 1$ , we modify the MS in eq. (32) as

$$\begin{aligned} u &= (y - 1) \sin(2\pi x) \cos(2\pi y) \\ v &= -(y - 1) \sin(2\pi y) \cos(2\pi x) \\ p &= (y - 1) (\cos(4\pi x) + \cos(4\pi y)) \end{aligned} \quad (40)$$

Clearly, at  $y = 1$  we have boundary values  $u = v = p = 0$ . In SPH, the Dirichlet boundary may be applied by setting the desired value of the property on the ghost layer shown in orange in fig. 16. We set homogenous velocity and pressure boundary conditions in two separate testcases and refer to them as *velocity BC* and *pressure BC*, respectively. We set the pressure/velocity on the solid using the MS when we set velocity/pressure using the SPH method. We simulate the problem for 100 timesteps with the MS in eq. (40).

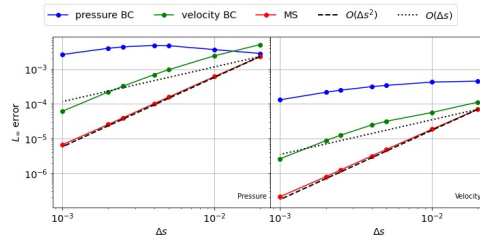


FIG. 17. The error in pressure (left) and velocity (right) with fluid particles initialized using the MS in eq. (40) 100 timesteps for L-IPST-C and *velocity BC* and *pressure BC* applied at the orange boundary in fig. 16.

In fig. 17, we plot the  $L_{\infty}$  error in pressure and velocity as a function of resolution for L-IPST-C, *velocity BC*, and *pressure BC*. Clearly, both the boundary conditions introduce error in the solution. The error introduced due to *Velocity BC* remains around second-order in pressure and first-order in velocity. The *pressure BC* is rarely used in SPH and introduces a significant amount of error with almost zero order convergence.

### 2. Slip boundary condition

In the SPH method, the slip boundary condition can be applied using the method proposed by Maciá *et al.*<sup>46</sup>. First, we extrapolate the velocity of the fluid to the solid using

$$\mathbf{u}_s = \frac{\sum \mathbf{u}_f W_{sf}}{\sum_j W_{sf}}, \quad (41)$$

where  $\mathbf{u}$ , and  $\mathbf{u}_f$  denotes the velocity of wall and fluid particles, respectively. Then, we reverse the component of the velocity normal to the wall. This method ensures that the divergence of velocity is captured accurately near the slip wall. Therefore, we consider the inviscid MS given by

$$\begin{aligned} u(x, y) &= (y - 1)^2 \sin(2\pi x) \cos(2\pi y) \\ v(x, y) &= -\sin(2\pi y) \cos(2\pi x) \\ p(x, y) &= (y - 1) (\cos(4\pi x) + \cos(4\pi y)) \end{aligned} \quad (42)$$

We note that the  $u$  velocity is symmetric across  $y = 1$  and  $v$  velocity is asymmetric. We consider the domain as shown in fig. 16 and apply the free slip boundary condition on the solid boundary shown in orange color for the L-IPST-C scheme. We refer to this method as *slip BC*. We note that the pressure and density on the solid is set using the MS. We simulate the problem for 100 timesteps. In fig. 18, we plot the  $L_1$  error in

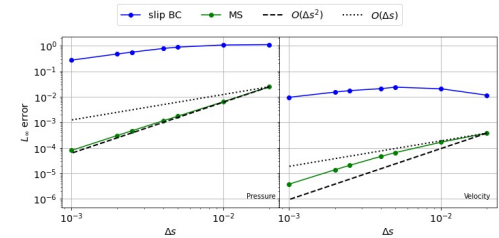


FIG. 18. The error in pressure (left) and velocity (right) with fluid particles initialized using the MS in eq. (42) after 100 timesteps for L-IPST-C and *slip BC* applied on the orange boundary in fig. 16.

pressure and velocity as a function of resolution for L-IPST-C and *slip BC* schemes. Clearly, the application of slip boundary condition increases the error and the order of convergence is less than one. In the case of the L-IPST-C scheme, the lower resolutions show first order convergence but as the resolution increases approaches second-order. We note that the fig. 18 shows the  $L_{\infty}$  error, however convergence of the  $L_1$  error is close to second-order for all resolutions. In summary, the slip boundary condition as proposed in 46 is accurate in velocity but reduces the accuracy of the pressure.

### 3. Pressure boundary condition

In the pressure boundary condition proposed by Maciá *et al.*<sup>46</sup>, we ensure that the pressure gradient normal to the

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 14

boundary is zero. We apply the boundary condition by setting  
the pressure of the solid boundary particles using

$$p_s = \frac{\sum p_f W_{sf}}{\sum_j W_{sf}}, \quad (43)$$

where  $p_s$  and  $p_f$  denotes the pressure of wall and fluid particles, respectively. For simplicity, we ignore the acceleration due to gravity and motion of the solid body. We consider the MS of the form

$$\begin{aligned} u(x,y) &= y^2 \sin(2\pi x) \cos(2\pi y) \\ v(x,y) &= -\sin(2\pi y) \cos(2\pi x) \\ p(x,y) &= (y-1)^2 (\cos(4\pi x) + \cos(4\pi y)) \end{aligned} \quad (44)$$

Clearly, the MS satisfies  $\frac{\partial p}{\partial y} = 0$  at  $y = 1$ . We consider the domain as shown in fig. 16 and apply the pressure boundary condition on the solid boundary shown in orange color for L-IPST-C scheme. We refer to this method as *Neumann BC*. We simulate the problem for 100 timesteps.

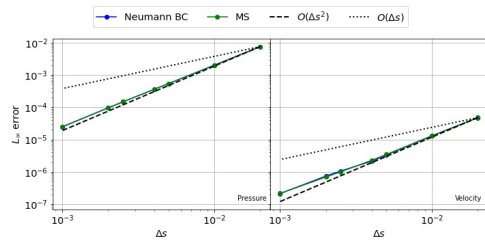


FIG. 19. The error in pressure (left) and velocity (right) with fluid particles initialized using the MS in eq. (44) after 100 timesteps for L-IPST-C and *Neumann BC* applied on the orange boundary in fig. 16.

In fig. 19, we plot the  $L_\infty$  error in pressure and velocity for L-IPST-C and *Neumann BC*. The results show that the pressure boundary condition is second order convergent.

#### 4. No-slip boundary condition

Maciá *et al.*<sup>46</sup> proposed the no-slip boundary condition for SPH where we set the wall velocity as

$$\mathbf{u}_s = 2\mathbf{u}_w - \tilde{\mathbf{u}}_s, \quad (45)$$

where  $\mathbf{u}_w$  is velocity of the wall and  $\tilde{\mathbf{u}}_s$  is the Shepard interpolated velocity (see eq. (41)). In the no-slip boundary, we ensure that  $\frac{\partial u}{\partial y} = 0$  at  $y = 1$  therefore, we use the MS for viscous flow given by

$$\begin{aligned} u(x,y,t) &= (y-1)^2 e^{-10t} \sin(2\pi x) \cos(2\pi y) \\ v(x,y,t) &= -(y-1)^2 e^{-10t} \sin(2\pi y) \cos(2\pi x) \\ p(x,y,t) &= (\cos(4\pi x) + \cos(4\pi y)) e^{-10t} \end{aligned} \quad (46)$$

We consider the domain as shown in fig. 16 and apply the pressure boundary condition on the solid boundary shown in orange color for the L-IPST-C scheme. We refer to this method as *no-slip BC*. We simulate the problem for 100 timesteps with  $v = 1.0m^2/s$ .

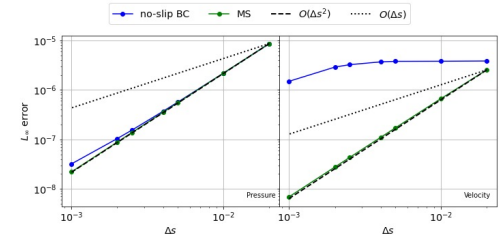


FIG. 20. The error in pressure (left) and velocity (right) with fluid particles initialized using the MS in eq. (44) after 100 timesteps for L-IPST-C and *no-slip BC* applied on the orange boundary in fig. 16.

In fig. 20, we plot the  $L_\infty$  error in pressure and velocity for 100 timesteps. Clearly, the *no-slip BC* shows increased error and a zero-order convergence. However, it does not introduce error in the pressure.

Thus in this section, we have demonstrated the MMS for obtaining the order of convergence of boundary condition implementations in SPH.

#### G. Convergence and extreme resolutions

Thus far we have used particle resolutions in the range  $10^{-3} \leq \Delta s \leq 2 \times 10^{-2}$ . We wish to study the convergence of the scheme when much higher resolutions are considered. We consider a domain of size  $1 \times 1$  with uniformly distributed particles as shown in fig. 21. In order to reduce computation, we reduce the size of the domain by half if the number of particles crosses  $1M$ . In the fig. 21, the red box shows the domain considered for the computation which one million particles with  $\Delta s = 1.25 \times 10^{-4}$ . In order to obtain an unbiased error estimate we consider same MS and the domain shown by black box in fig. 21 to evaluate  $L_\infty$  error using eq. (39).

We first consider the MS given in eq. (30). We solve the eq. (23) using the L-IPST-C scheme for all the resolutions with  $v = .01m^2/s$ . We consider the case where we do not correct the kernel gradient in the discretization of eq. (23) in the L-IPST-C scheme.

In fig. 22, we plot the error in pressure and velocity solved using L-IPST-C scheme with kernel gradient corrected, after 100 timesteps as a function of resolution for  $h_{\Delta s} = 1.2$  and  $h_{\Delta s} = 1.4$ . Clearly, We obtain second order convergence. In fig. 23, we plot the error for the case where we do not employ kernel gradient correction. Clearly, the discretization error dominates.

We also consider the MS containing a range of frequencies

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How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 15

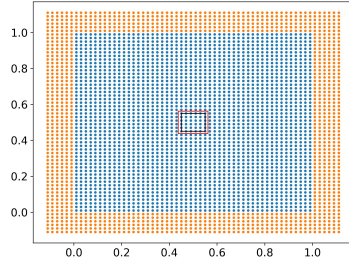


FIG. 21. The domain filled by blue fluid particles. The red box shows the smallest domain considered for the highest resolution of  $8000 \times 8000$  and the black box shows the area which is considered to evaluate error for all the resolutions.

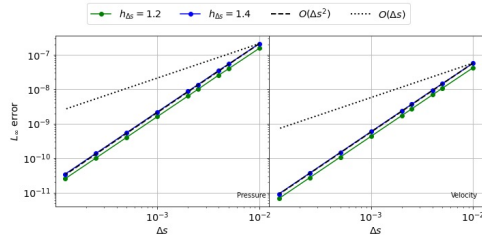


FIG. 22. The error in pressure (left) and velocity (right) as a function of resolution for two different  $h_{\Delta s}$  values with the MS in eq. (30). All cases are solved using L-IPST-C scheme with kernel gradient correction.

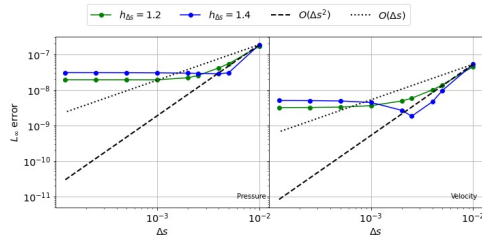


FIG. 23. The error in pressure (left) and velocity (right) as a function of resolution for two different  $h_{\Delta s}$  values with the MS in eq. (30). All cases are solved using L-IPST-C scheme with no kernel gradient correction.

given by

$$u(x, y, t) = y^2 e^{-10t} \sum_{j=1}^{10} \sin(2j\pi x) \cos(2j\pi y)$$

$$v(x, y, t) = -e^{-10t} \sum_{j=1}^{10} \sin(2j\pi y) \cos(2j\pi x) \quad (47)$$

$$p(x, y, t) = e^{-10t} \sum_{j=1}^{10} \cos(4j\pi x) + \cos(4j\pi y).$$

We simulate the eq. (6) using L-IPST-C scheme for the above MS. As before, we also consider the case where we do not employ kernel correction.

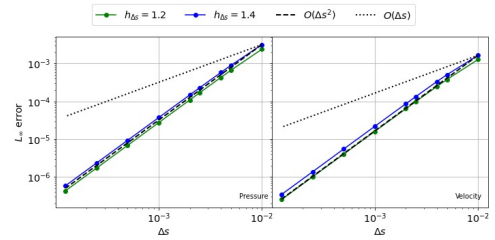


FIG. 24. The error in pressure (left) and velocity (right) as a function of resolution for two different  $h_{\Delta s}$  values with the MS in eq. (47). All cases are solved using L-IPST-C scheme with kernel gradient correction.

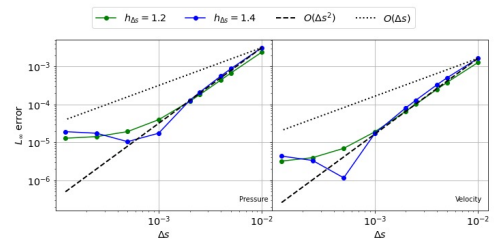


FIG. 25. The error in pressure (left) and velocity (right) as a function of resolution for two different  $h_{\Delta s}$  values with the MS in eq. (47). All cases are solved using L-IPST-C scheme with no kernel gradient correction.

In fig. 24, we plot the error in pressure and velocity solved using L-IPST-C scheme with kernel gradient correction for 100 timesteps as a function of resolutions. Clearly, both the cases shows second-order convergence. In fig. 25, we plot the error in pressure and velocity for the solution obtained using L-IPST-C scheme with no kernel correction. As can be seen the kernel correction is essential in order to obtain second-order convergence at high resolutions.

We have therefore shown that we can consider very high resolutions using the MMS technique. This enables us to find

How to train your solver: A method of manufactured solutions for weakly-compressible smoothed particle hydrodynamics 16

967 flaws in the scheme which may not converge at very high reso-995  
968 lution. These are hard to test using traditional methods where996  
969 an actual problem is solved. 997

## 970 H. Verification in 3D 1000

971 We now use the MMS to verify a three dimensional solver992  
972 Since the number of particles in three-dimensions increases993  
973 much faster than in two-dimensions, we can reduce the do-1004  
974 main size with resolution as done while dealing with extreme995  
975 resolutions. We consider a unit cube domain size with 1 mil-1006  
976 lion particles. As we increase the resolution, we decrease the997  
977 size of the domain such that the number of particles in the998  
978 domain remains at 1 million. We consider the MS given by 1009

$$\begin{aligned} u(x, y, z, t) &= y^2 e^{-10t} \sin(\pi(2x + 2z)) \cos(\pi(2x + 2y)) & 1010 \\ v(x, y, z, t) &= -e^{-10t} \sin(\pi(2y + 2z)) \cos(\pi(2x + 2y)) & 1011 \\ w(x, y, z, t) &= -e^{-10t} \sin(\pi(2x + 2z)) \cos(\pi(2y + 2z)) & 1012 \\ p(x, y, z, t) &= (\cos(\pi(4x + 4y)) + \cos(\pi(4x + 4z))) e^{-10t} & 1013 \end{aligned}$$

980 We obtain the source term by subjecting the MS in eq. (48)1017  
981 to the governing equation in eq. (6) with  $\nu = 0.01m^2/s$ . We1018  
982 simulate the problem for 10 timesteps. 1019

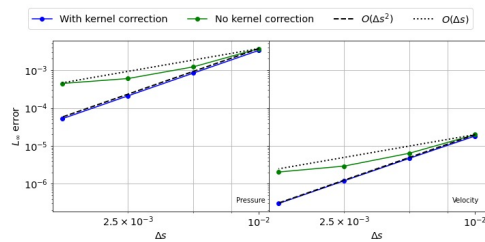


FIG. 26. The  $L_\infty$  error in pressure (left) and velocity (right) after 10 timesteps as a function of resolution solved using L-IPST-C scheme with and without kernel correction. The source term are calculated using the MS in eq. (48).

983 In fig. 26, we plot the  $L_\infty$  error in pressure and velocity as993  
984 a function of resolution for L-IPST-C scheme with and with994  
985 out kernel correction. As expected, the case with no kernel995  
986 correction gradually flatten due dominance of discretization996  
987 error. The case with kernel correction shows second order997  
988 convergence in both pressure and velocity. Thus we see that998  
989 we can easily test the SPH method in a three-dimensional do-1004  
990 main using the MMS. 1042

## 991 VI. DISCUSSION 1043

992 We have used the MMS to verify the convergence of differ-1047  
993 ent WCSPH schemes. Thus far, most of the numerical stud-1048  
994 ies of the accuracy and convergence of the WCSPH method1049

have used either an exact solution like the Taylor-Green vortex problem, or with an established solver, or experimental result. These methods are therefore limited in their ability to detect specific problems in an SPH implementation. This is true even in the recent work of Negi and Ramachandran<sup>24</sup> where a Taylor-Green problem and a Gresho-Chan vortex problem is used. These are complex problems and obtaining a solution to these involves a significant amount of computation. Moreover, if the results do not produce the expected accuracy or convergence, the researcher does not obtain much insight into the origin of the problem. Furthermore, the established approaches do not offer any means to study the accuracy of boundary condition implementations.

In this context, the proposed approach offers a multitude of advantages listed and discussed below:

- The method is highly efficient in terms of execution time. We are able to detect problems in the implementations of specific discretization operators in less than 100 iterations. Even for our most challenging cases with a million particles, the typical run time for a single computation on a multi-core CPU does not exceed a few minutes. On the other hand, the comparison study for the lid-driven cavity case in section III took 150 minutes for the  $200 \times 200$  resolution.
- The method easily works in three dimensions and we demonstrate its applicability for a simple three-dimensional case. This is significant because traditional SPH verifications only use two-dimensional problems.
- We can effectively test the boundary condition implementations through this method. In this work we have demonstrated this for Dirichlet and Neumann boundary conditions in both pressure and velocity.
- The method allows us to identify very specific problems with a solver. Through a judicious choice of MS and time integrator, we can identify if the implementation of a specific governing equation is the source of a problem. We have demonstrated this with several examples in the preceding sections.
- We are able to verify the order of convergence efficiently even for very high resolutions and thereby test if the scheme is truly second order convergent as the resolution increases. In the present work we have demonstrated this for extremely high resolutions (involving  $8000 \times 8000$  particles) without needing to simulate the problem for a long duration and also limiting the number of computational particles to a smaller number.
- The method will work on any manufactured solution and this allows us to test the scheme with functions involving a large range of frequencies. In contrast, many exact solutions involve simple functional forms. Therefore by using the MMS the solver can be tested with a more challenging class of problems.

As a result of these significant advantages, the proposed method offers a robust, efficient, and powerful method to verify the accuracy and convergence of SPH schemes.



## 21050 VII. CONCLUSIONS 11003

21051 In this paper we propose the use of the method of manufactured solutions (MMS) in order to verify an SPH solver. While the MMS technique is well established in the context of mesh-based methods<sup>7</sup>, to the best of our knowledge it does not appear to have been employed in the context of Lagrangian SPH schemes thus far. The application of MMS to Lagrangian SPH method is non-trivial as the particles move.

21052 In the present work we show for the first time how the method can be employed to verify the accuracy of any modern weakly-compressible SPH scheme. Specifically, we note that for successful application of the MMS, quantities like gradient of velocity should be evaluated using the scheme and not with the gradient of the MS. In this paper, we apply PST to restrict the particles to remain inside the domain boundaries allowing us to apply MMS to arbitrary shaped boundaries without the need for addition and deletion of particles. We compare different initial particle distributions used in SPH to obtain a minimum number of iterations required for a result independent of initial distribution. We also show that one should not use a divergence free velocity field while using MMS in SPH for verification. We compare the EDAC and the PE-IPST-C schemes and show that the density should be used as a property independent of the neighbor particle distribution. We show that the method works in arbitrary number of dimensions, allowing us to systematically identify problems quickly in specific discretizations employed by the scheme, and makes it possible to verify the accuracy of boundary condition implementations as well. We also demonstrate that the recently proposed family of second order convergent WCSPH schemes<sup>24</sup> are indeed second order accurate. Finally, our implementation is open source ([https://gitlab.com/pypr/mms\\_sph](https://gitlab.com/pypr/mms_sph)) and our numerical experiments and results presented are fully automated in the interest of reproducibility. Given that convergence and accuracy of SPH schemes is a grand-challenge problem in the SPH community<sup>6</sup>, the present work offers a valuable contribution.

21053 In the future, we propose to use this method to study the accuracy and convergence of the method in the context of the various solid boundary conditions proposed in SPH. Using this method in the context of inlet and outlet boundary conditions and for free-surfaces may prove challenging and remain to be explored. The method may also be applied in the context of incompressible SPH, compressible SPH, and multi-phase SPH schemes. We plan to explore these problems in the future.

## 21054 ACKNOWLEDGMENTS 11058

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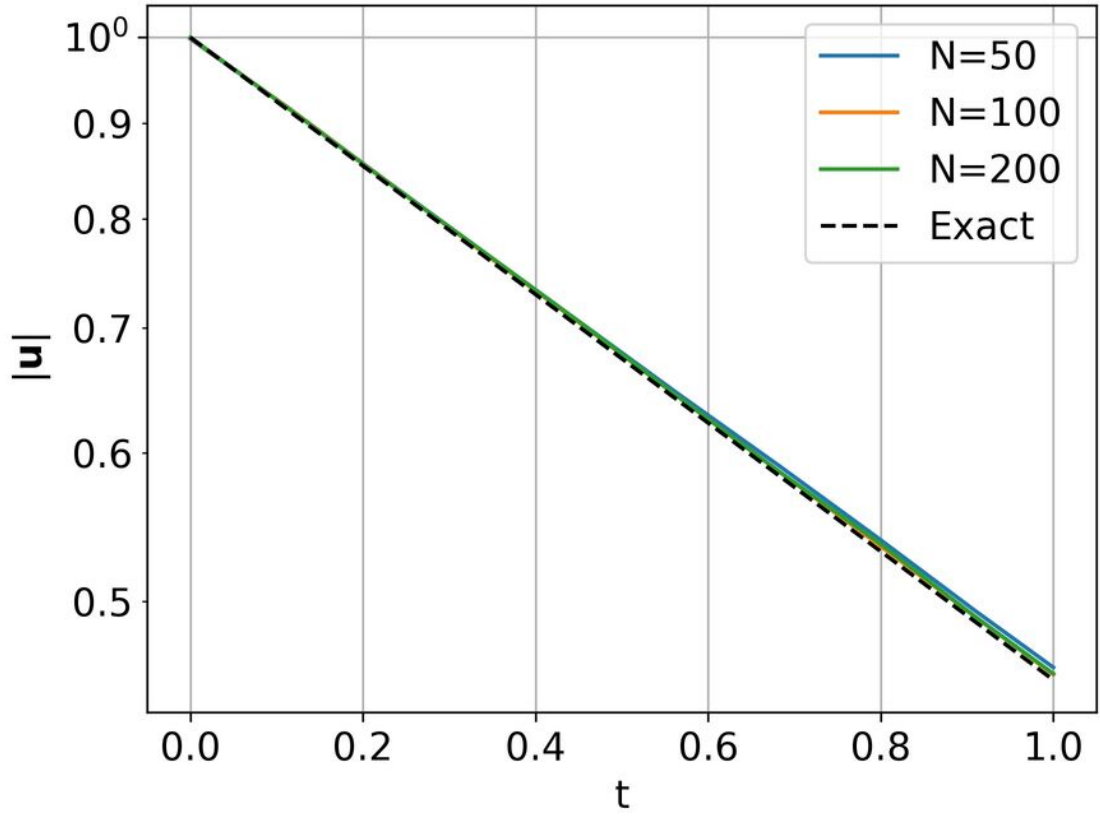
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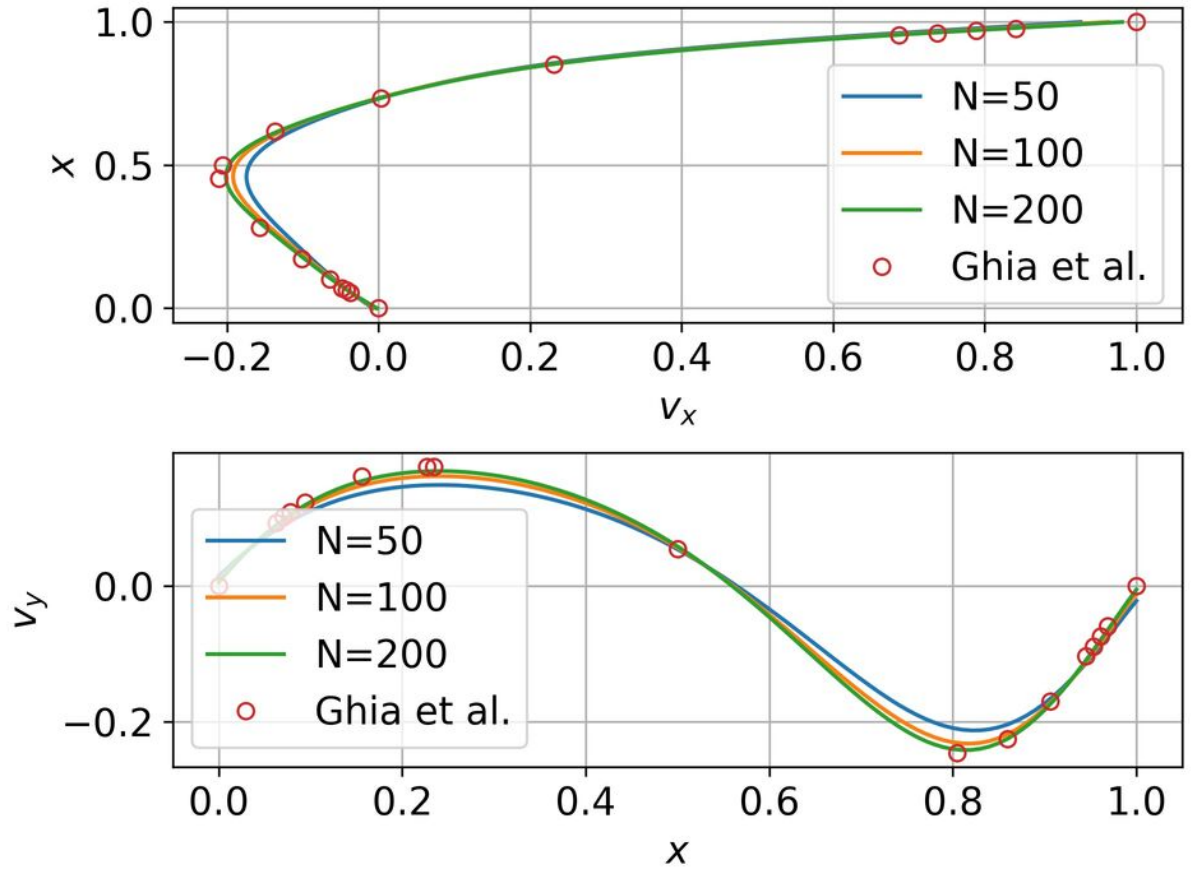
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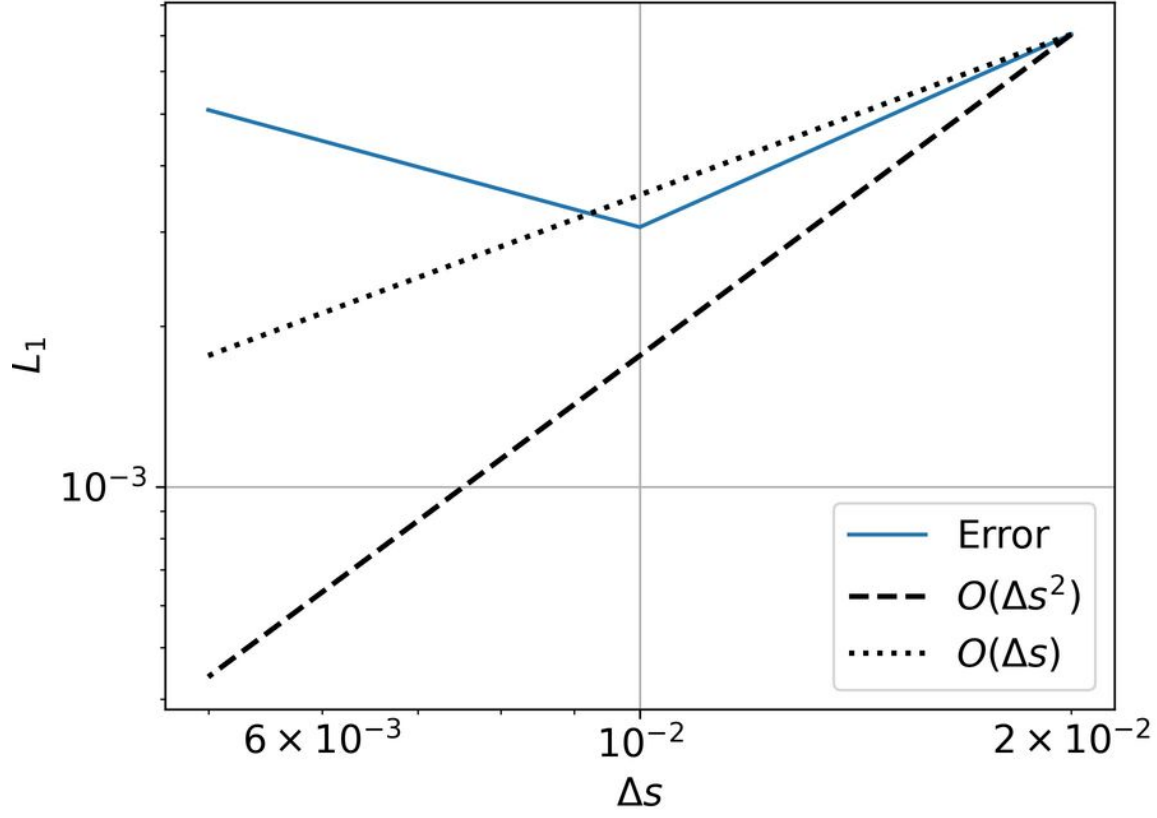
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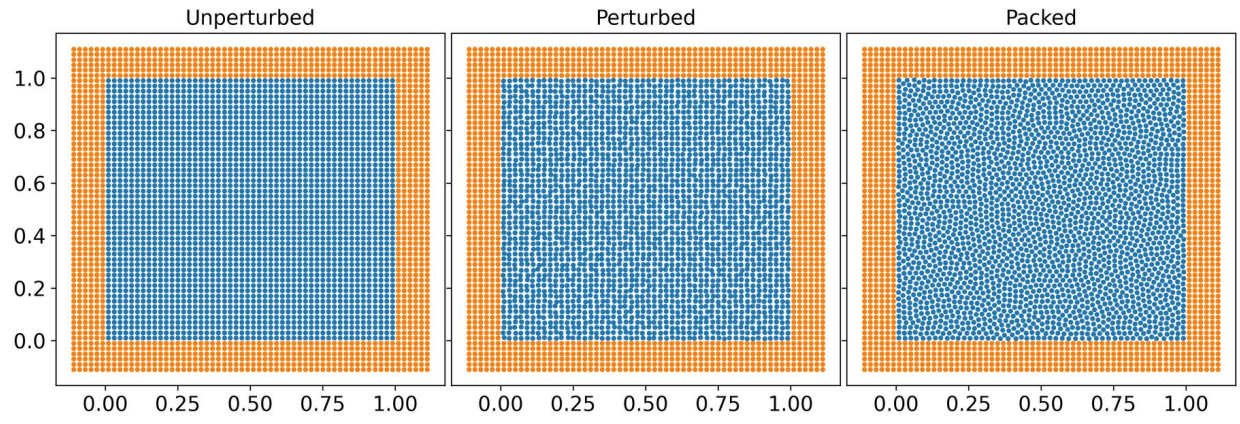
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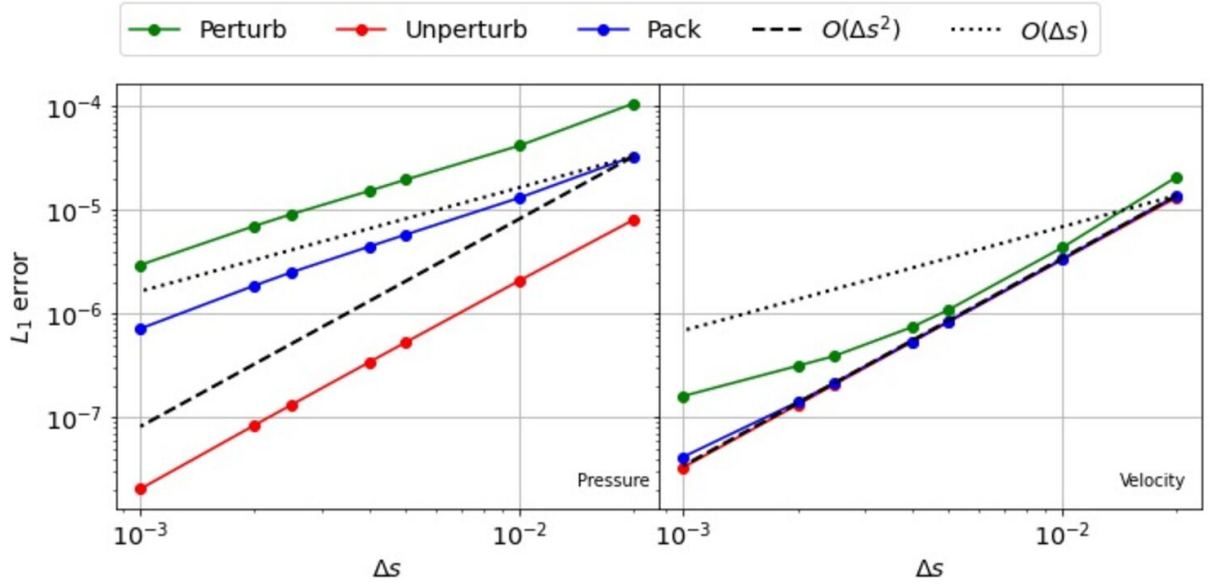


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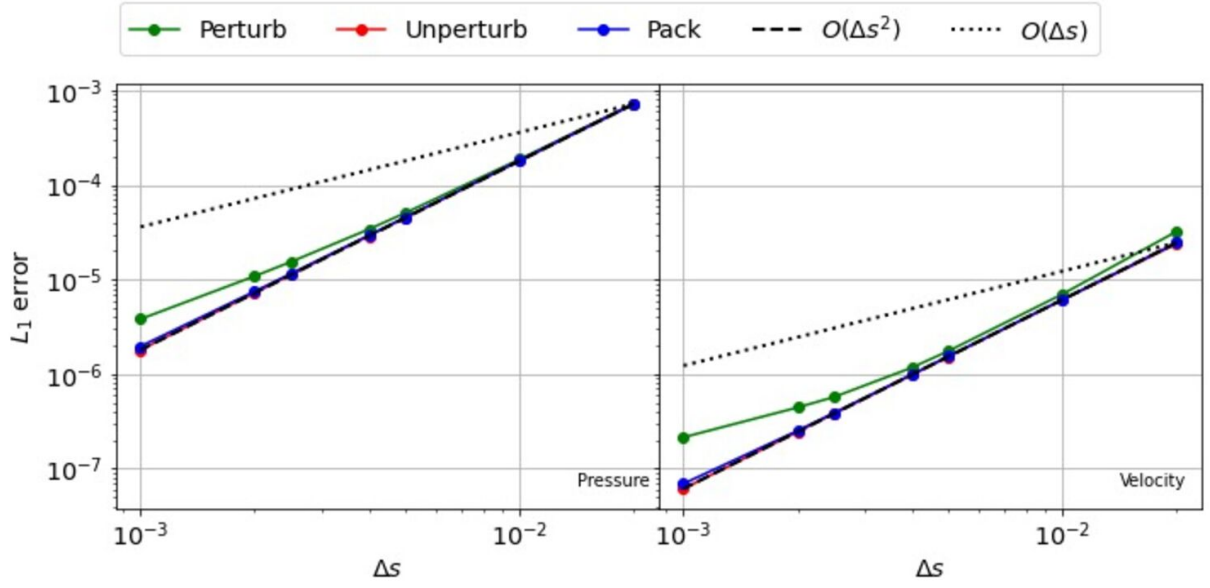






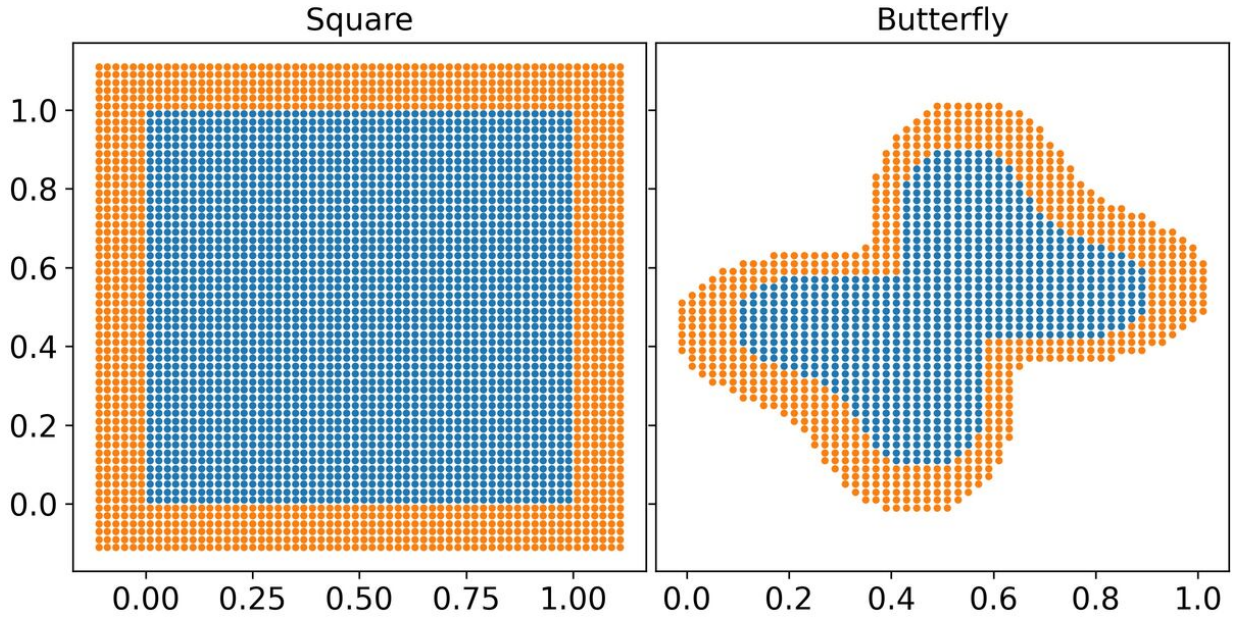
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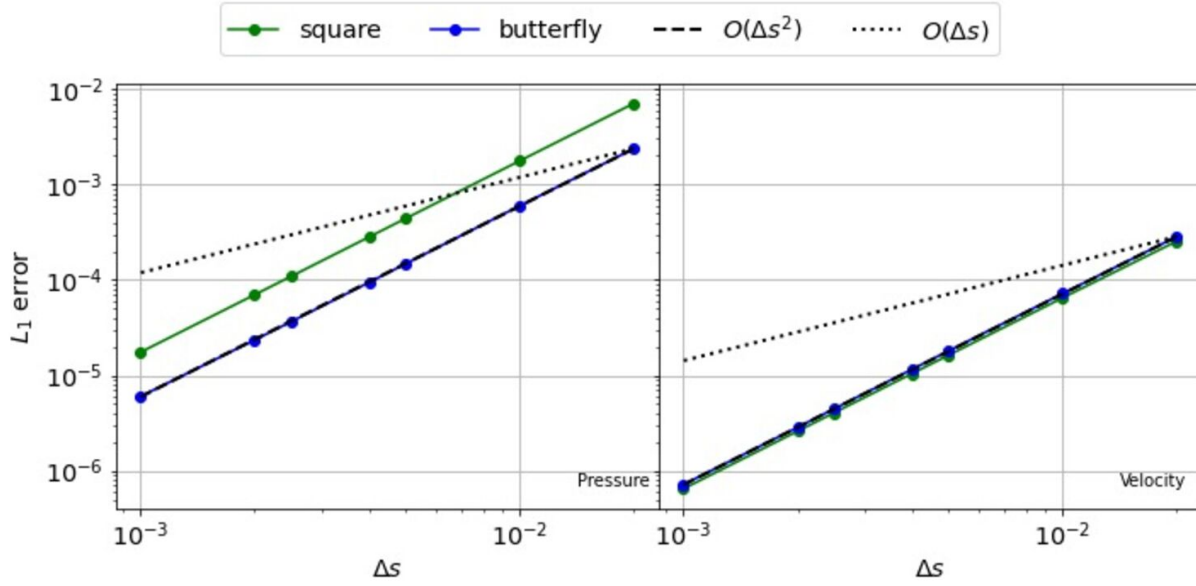
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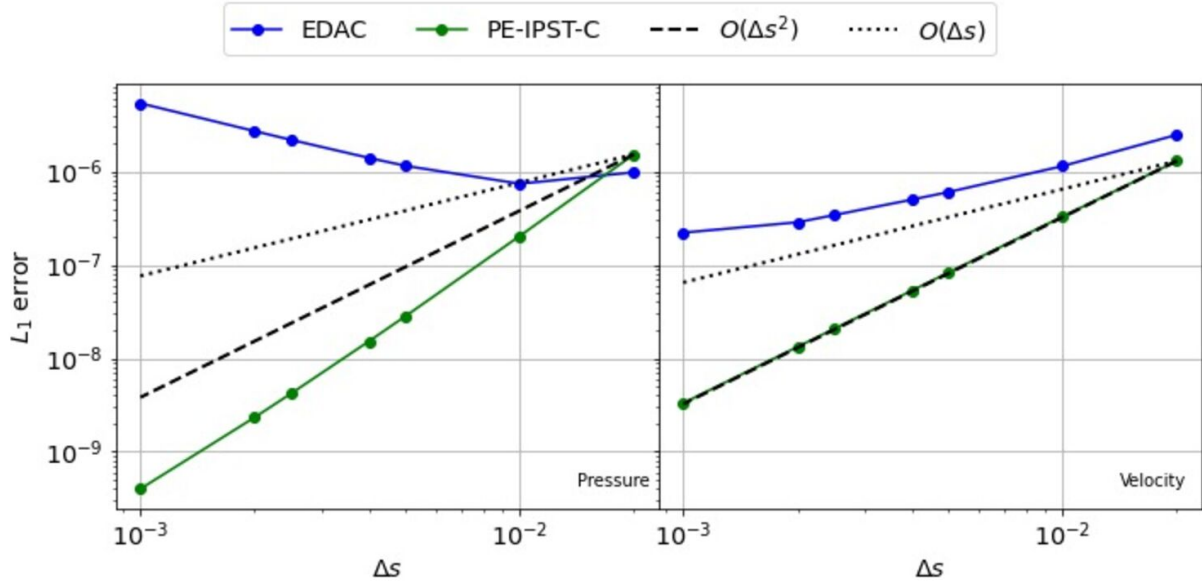
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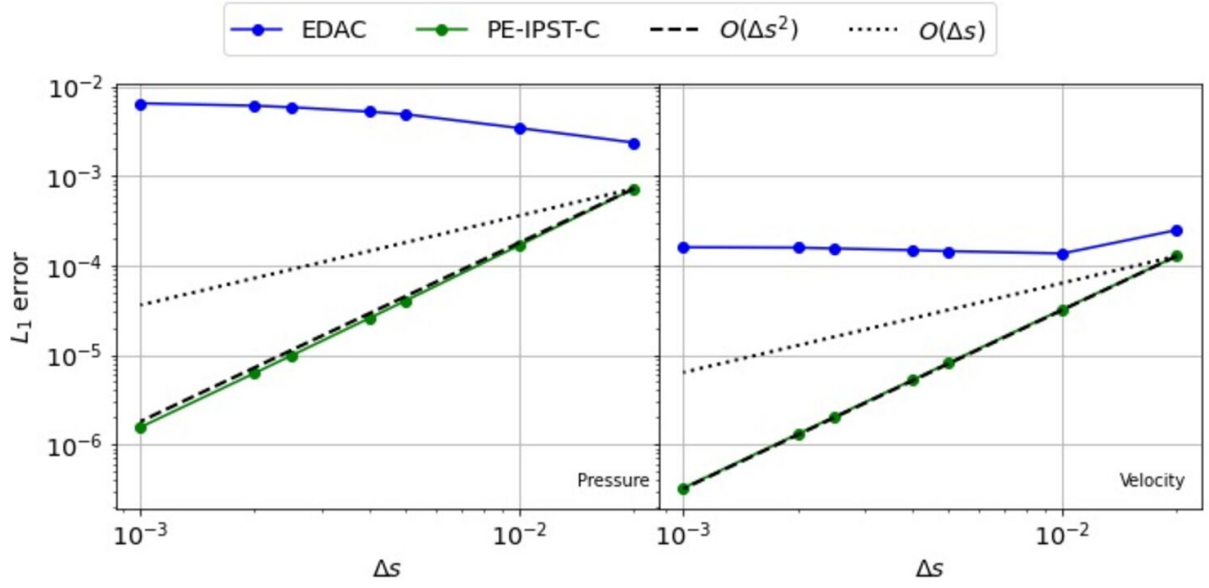
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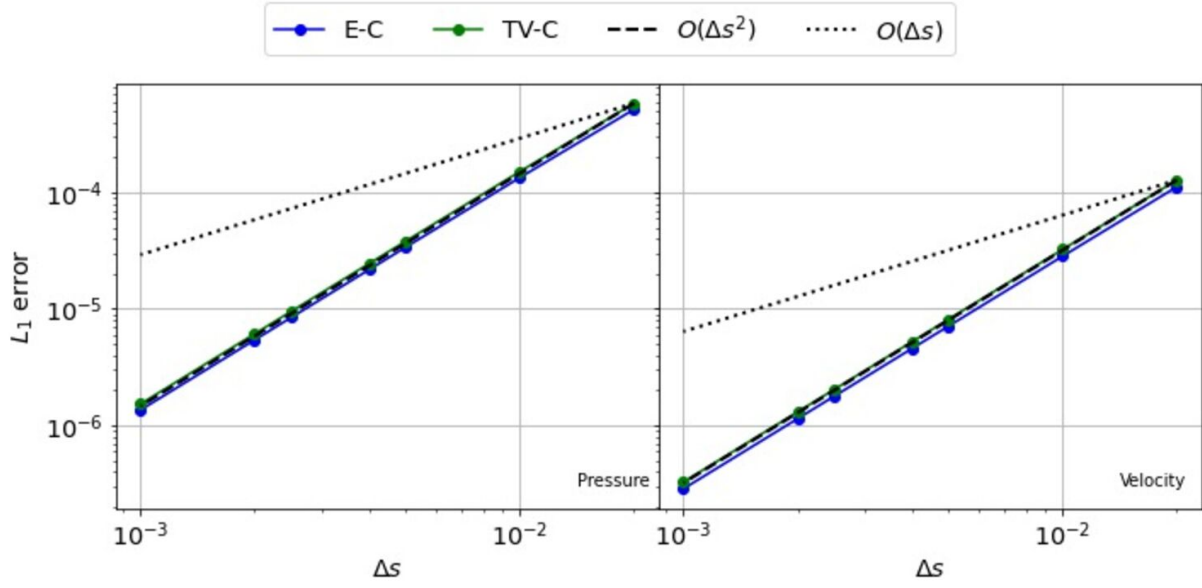
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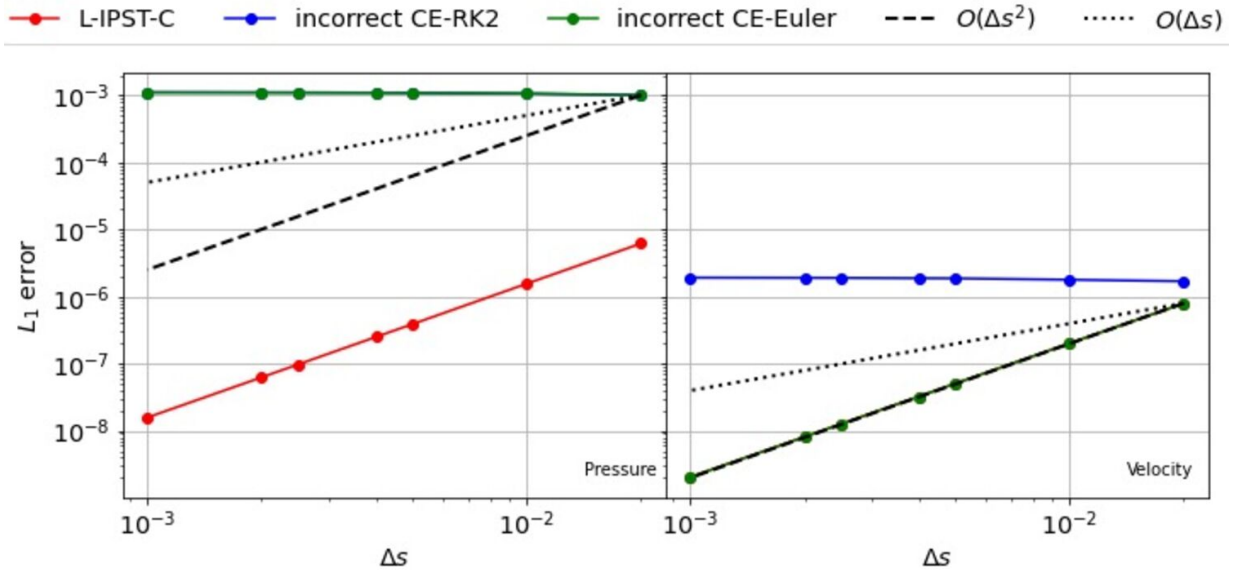
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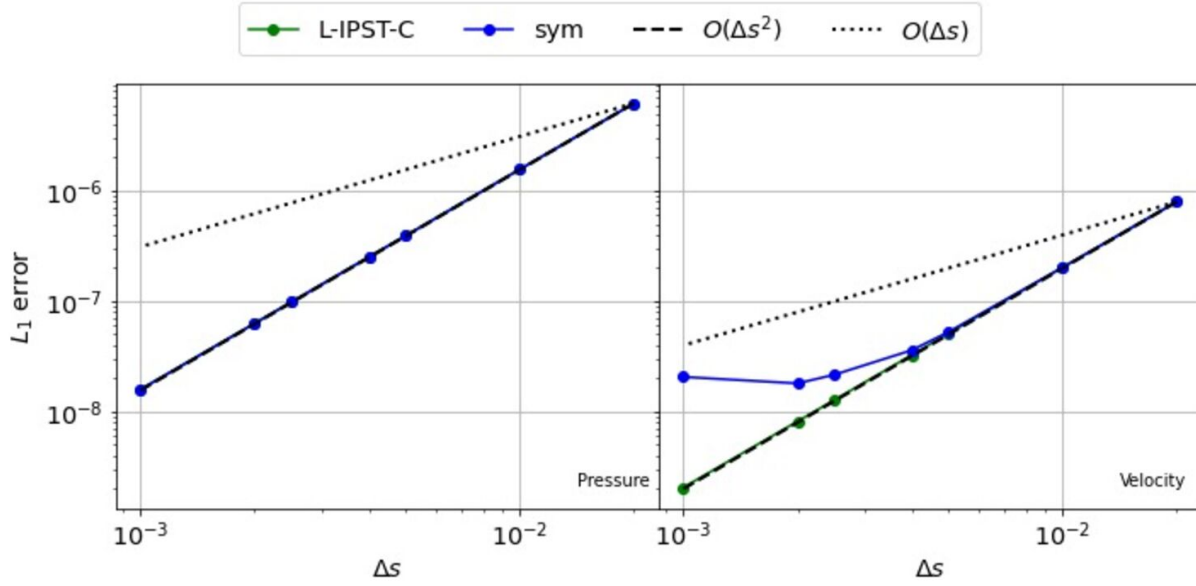
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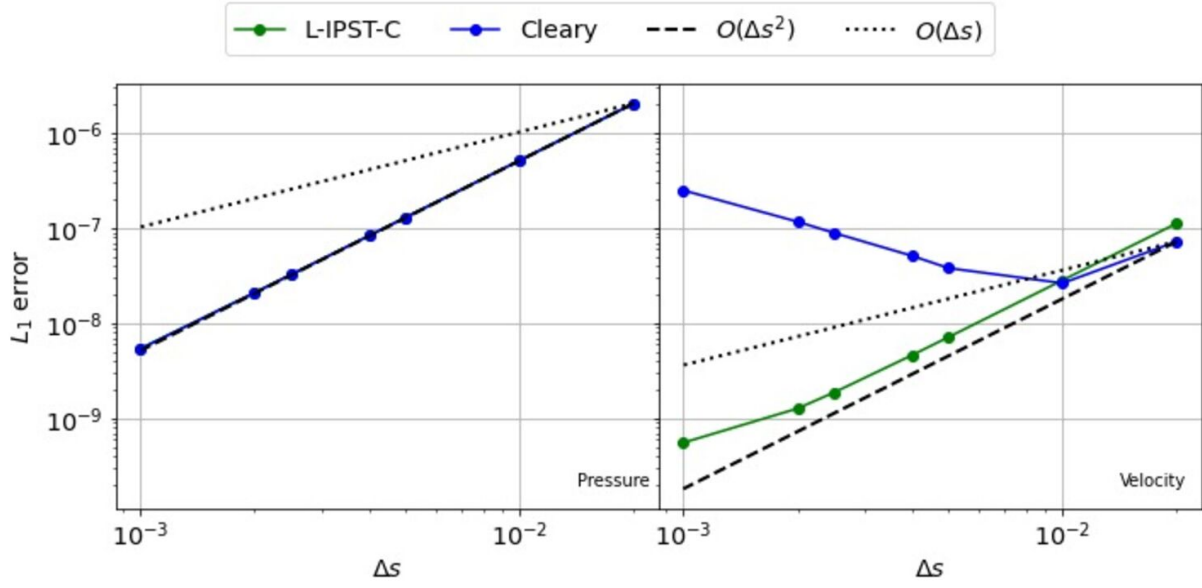
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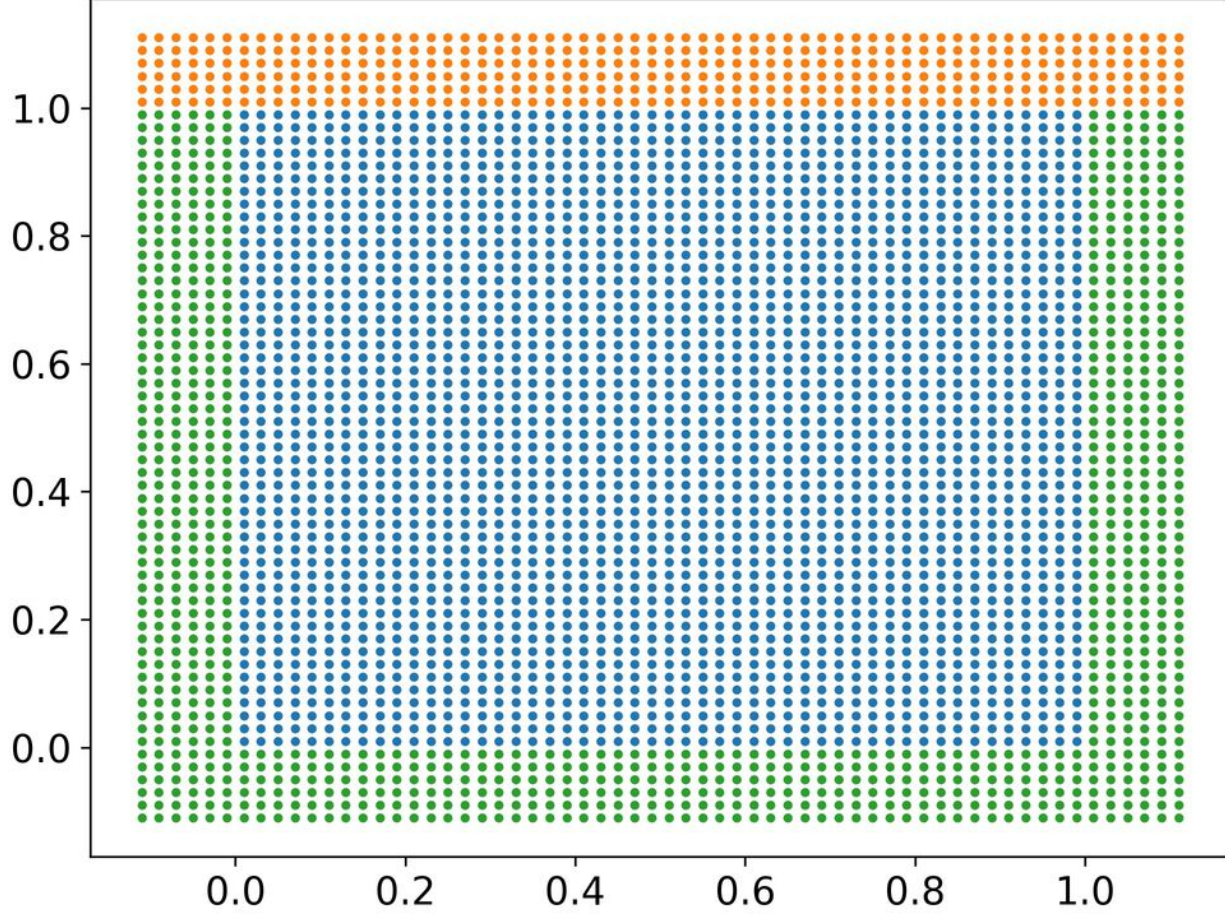
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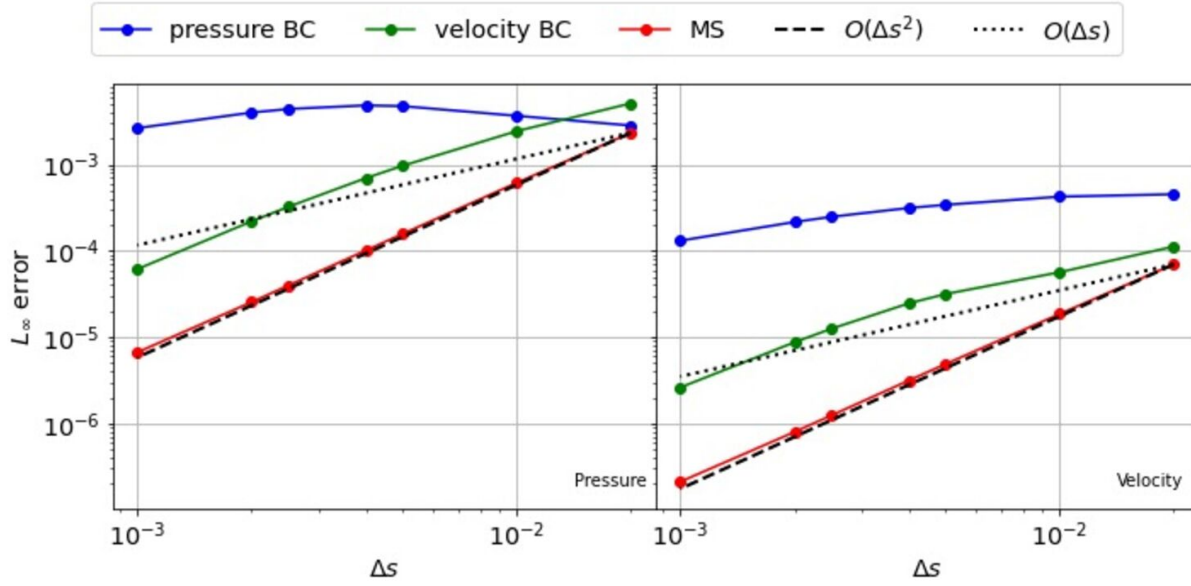
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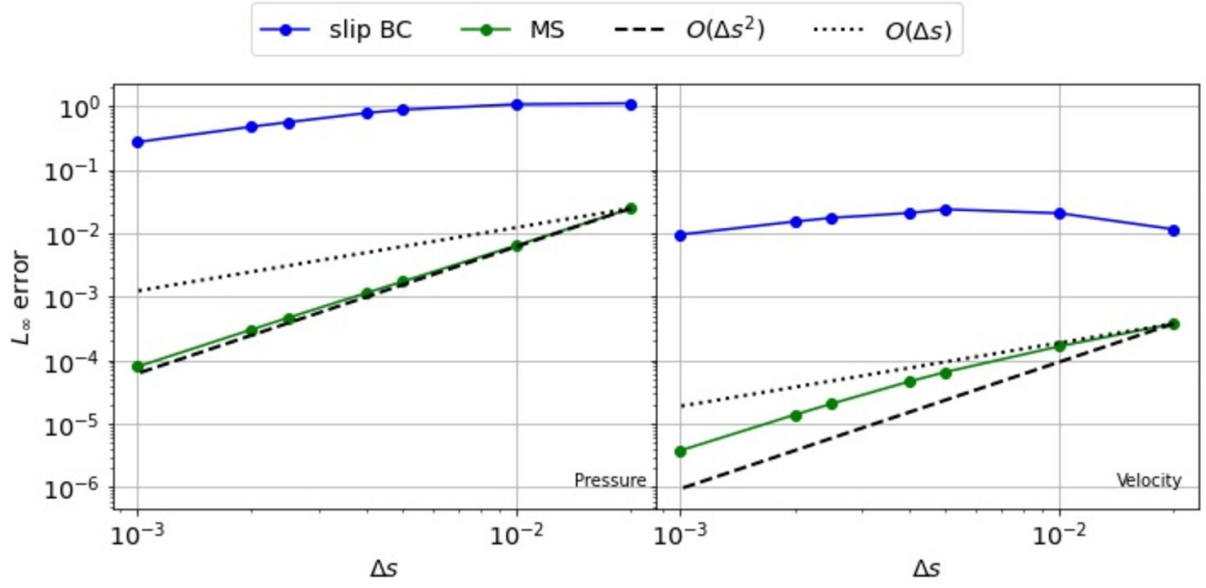
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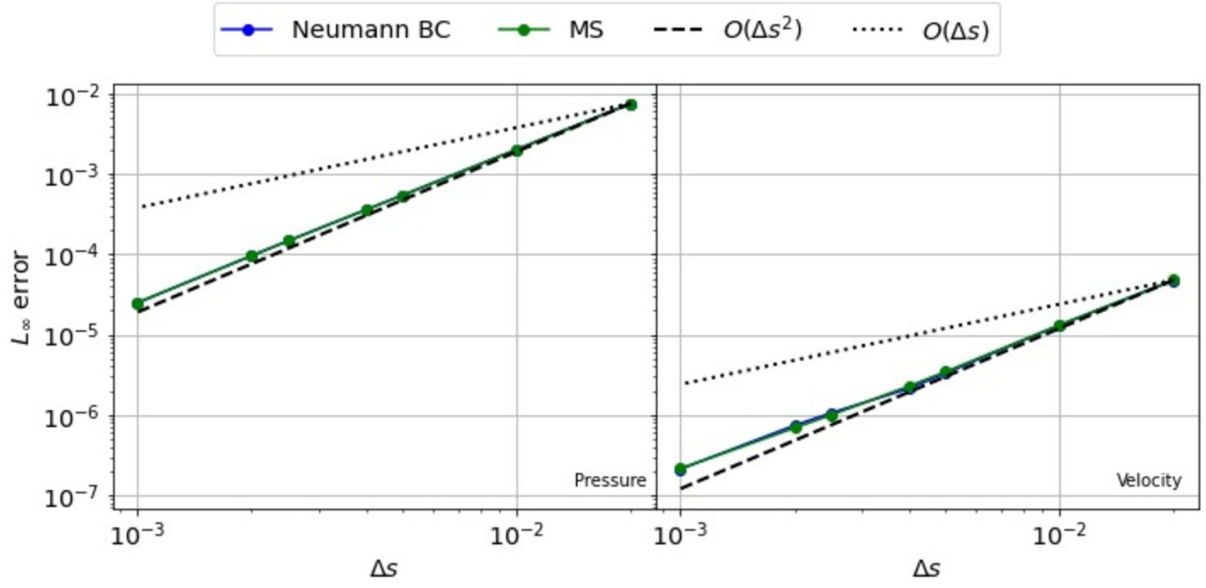
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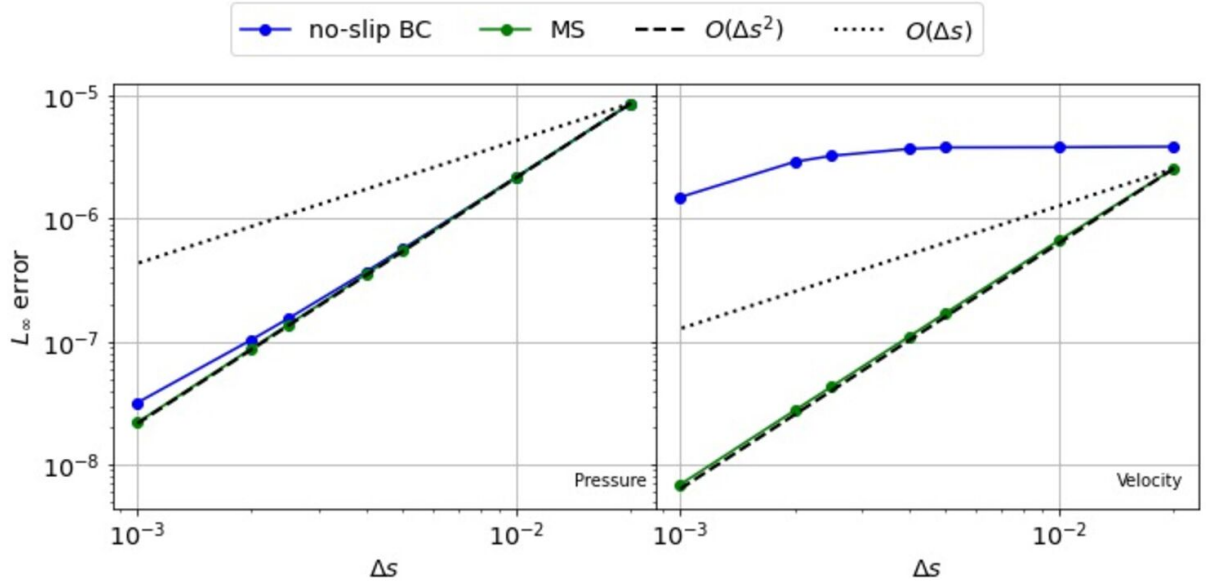
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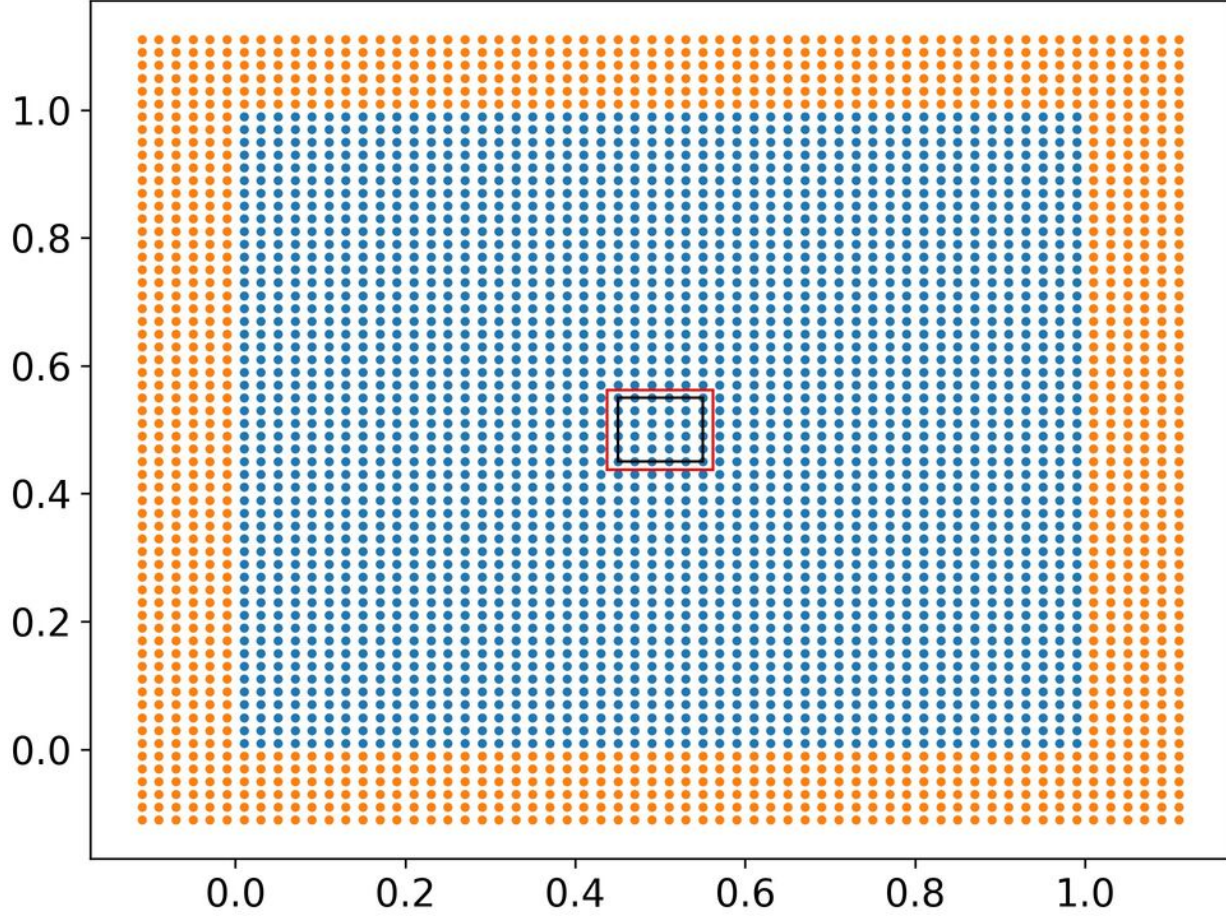
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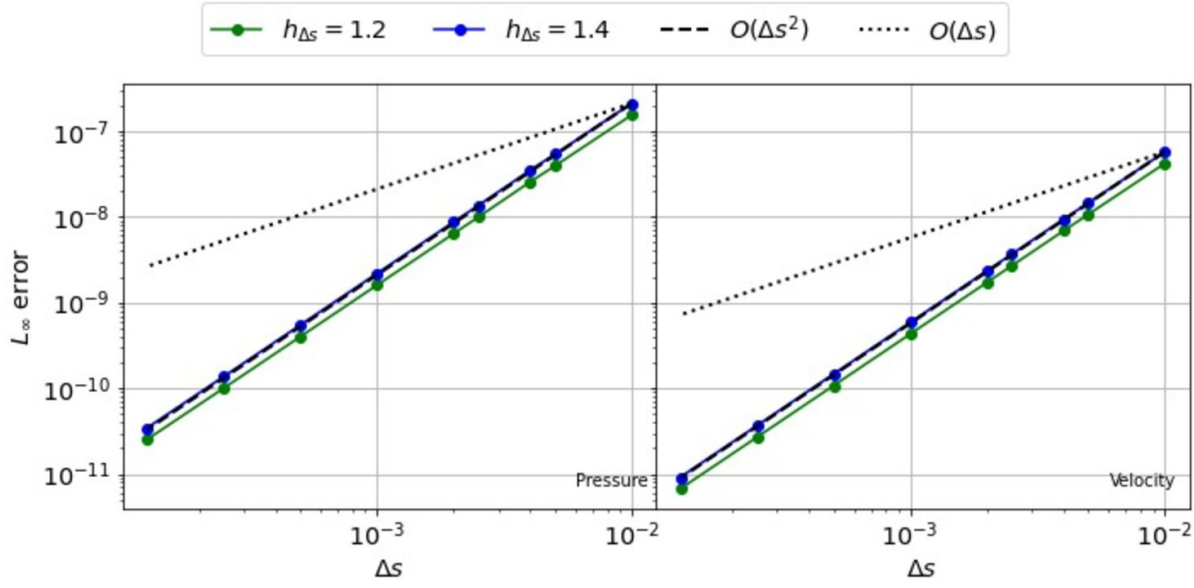
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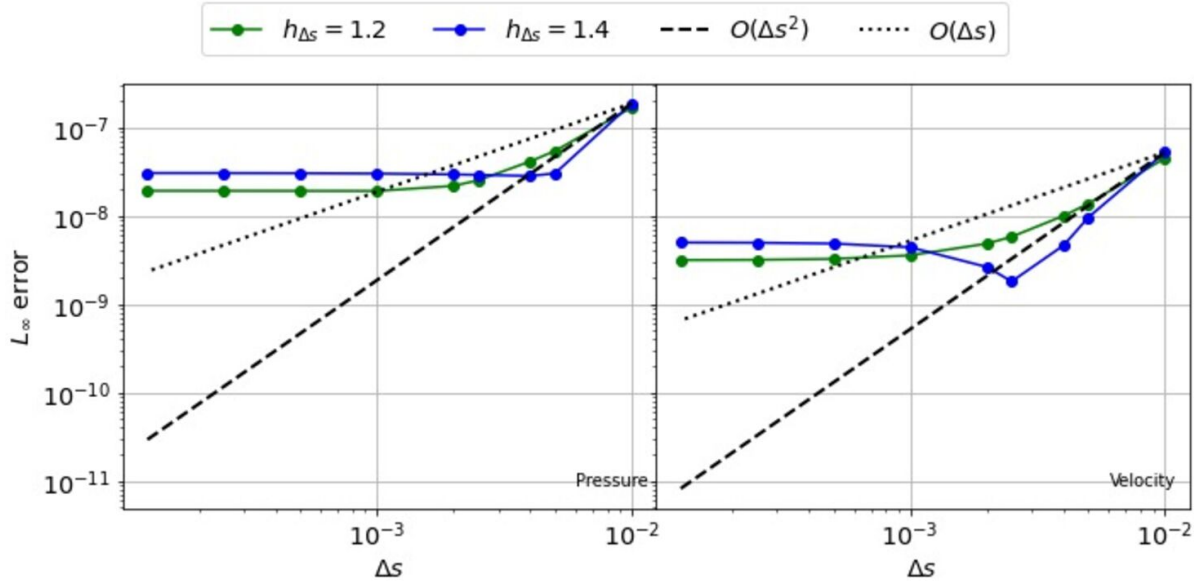
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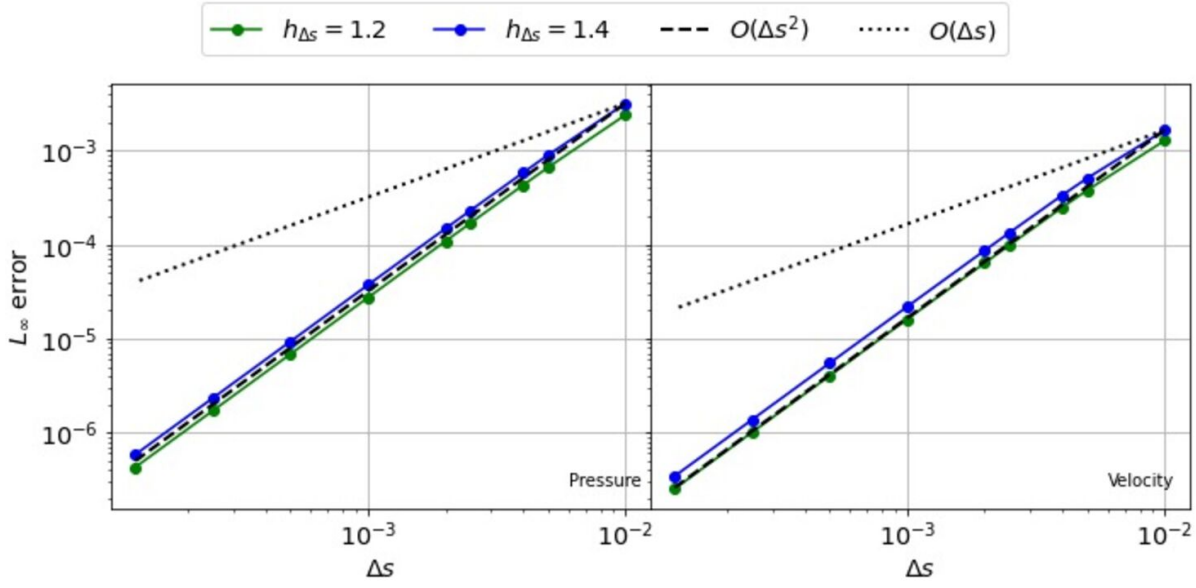
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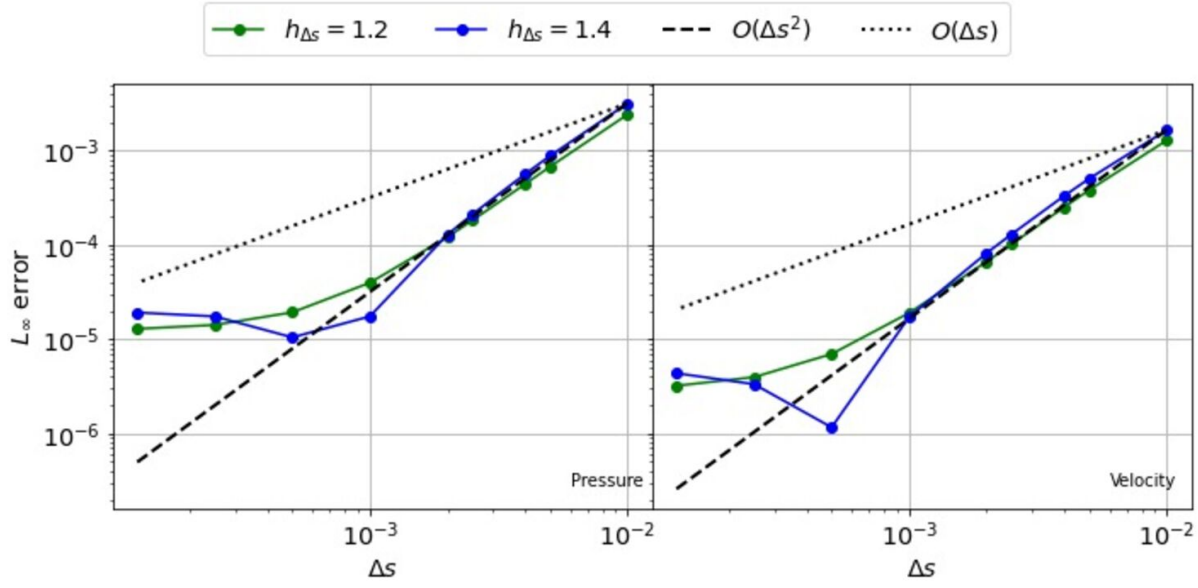
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